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# Electron Energy Dependent Charging Effects of Multilayered Dielectric Materials

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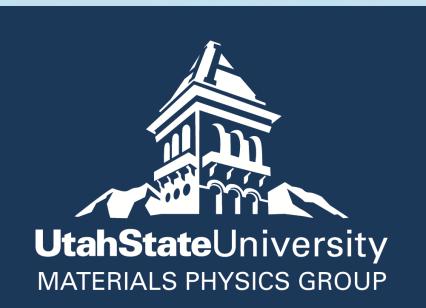
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# Abstract

Measurements of the charge distribution in electron-bombarded, thin-film, multilayered dielectric samples showed that charging of multilayered materials evolves with time and is highly dependent on incident energy; this is driven by electron penetration depth, electron emission and material conductivity. Based on the net surface potential's dependence on beam current, electron range, electron emission and conductivity, measurements of the surface potential, displacement current and beam energy allow the charge distribution to be inferred. To take these measurements, a thin-film disordered SiO<sub>2</sub> structure with a conductive middle layer was charged using 200 eV and 5 keV electron beams with regular 15 s pulses at 1 nA/cm<sup>2</sup> to 500 nA/cm<sup>2</sup>. Results show that there are two basic charging scenarios which are consistent with simple charging models; these are analyzed using independent determinations of the material's electron range, yields, and conductivity. Large negative net surface potentials led to electrostatic breakdown and large visible arcs, which have been observed to lead to detrimental spacecraft charging effects.

# Experimentation

In order to investigate the charging of multilayered dielectric materials, pulsed charging experiments were conducted using multilayered dielectric materials of an SiO<sub>2</sub> based optical Surface Voltage coating, a conductive middle layer and an SiO<sub>2</sub> substrate. Tests were made with the conductive layer both grounded and ungrounded. Experiments were conducted in the main USU electron emission ultrahigh vacuum test chamber, modified for observations of low intensity UV/VIS/NIR glow over a broad range of sample temperatures. Figure 1 provides a general schematic of the experimental system used.

The samples were subjected to short pulses  $(t_{on} \approx 15 \text{ s})$  of electron bombardment using a monoenergetic electron beam with beam energies of either 200 eV or 5 keV. A low energy electron gun [Staib, EK-5-S1] was used, that can deliver a well-characterized, low-flux pulsed beam (typically ~50pA/cm<sup>2</sup> to 1 µA/cm<sup>2</sup>) over an energy range of 20 eV to 5 the pulse charging surface voltage and electrode keV. The defocused electron beam produced a beam profile current data induced by electron beam bombardment. at the sample with about ±30% uniformity over an ~3 cm Instrumentation includes picoammeters, Pearson coils, diameter beam spot. Beam fluxes were monitored with a Faraday cup. Beam current densities of 20±1 nA/cm<sup>2</sup> at 200 eV and 2.7±1 nA/cm<sup>2</sup> at 5 keV were used for the experiment reported here, with an exposed sample area of 4.9±0.2 cm<sup>2</sup>.

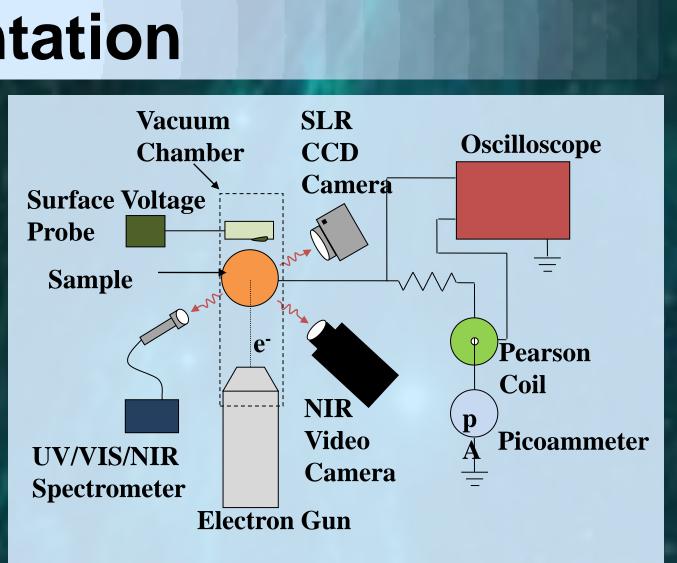


Fig. 1. Block diagram of instrumentation for collecting and a storage oscilloscope for electrode current measurements and UV/VIS and IR spectrometers, an SLR CCD still camera, and a NIR video camera for optical measurements.

Four experiments are considered as depicted in Fig. 6. The experiments differ in terms of the incident energy and flux, and as we will see below, produce dramatically different results. To interpret the experiments, we must consider three physical phenomena-the electron range, electron yield and the electron transport (conductivity) of the material—and how they are affected by the experimental conditions. <u>Range</u>

The electron range is the maximum distance an electron of a given incident energy can penetrate through a material at a given incident energy,  $E_{b}$ , as the incident electron undergoes a succession of energy loss collisions and ultimately deposits charge at  $R(E_{h})$  when all energy is expended (see Fig. 4). Figure 2(a) shows the results of a composite model for the energy dependence of the range spanning from a few eV to 10<sup>7</sup> eV. Knowing the range of electrons becomes especially critical when dealing with multilayered materials, where the incident energy will determine where and in what layer charge and energy are deposited. The low (200 eV) and high (5 keV) incident energies were selected for these experiments based on range calculations to deposit charge at the mid-point between the surface dielectric and the conductor and into the conductive layer, respectively

**Electron Yield** 

The total electron yield is defined as the ratio of emitted to incident flux and is highly energy dependent. The incident flux is the total number of electrons entering the material from the environment; the emitted flux is the sum of backscattered and secondary electrons, as shown in Fig. 4. Secondary electrons generally have energies <50 eV, while backscattered electrons generally have energies >50 eV.

# Theory

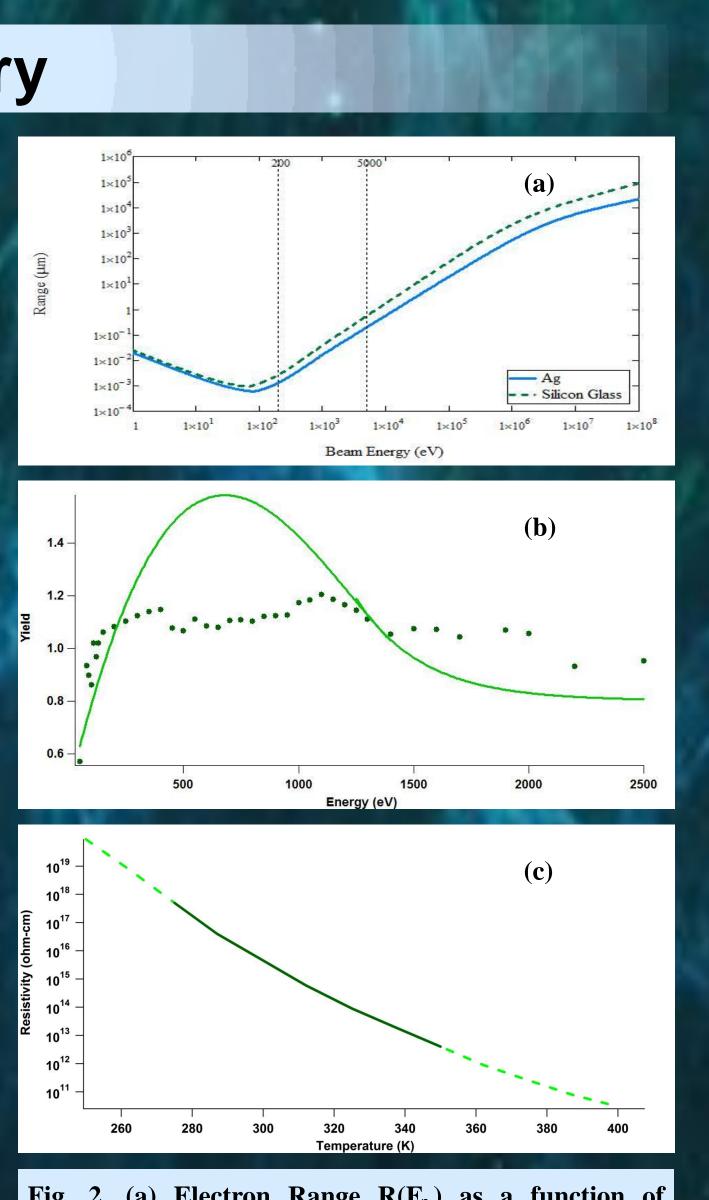
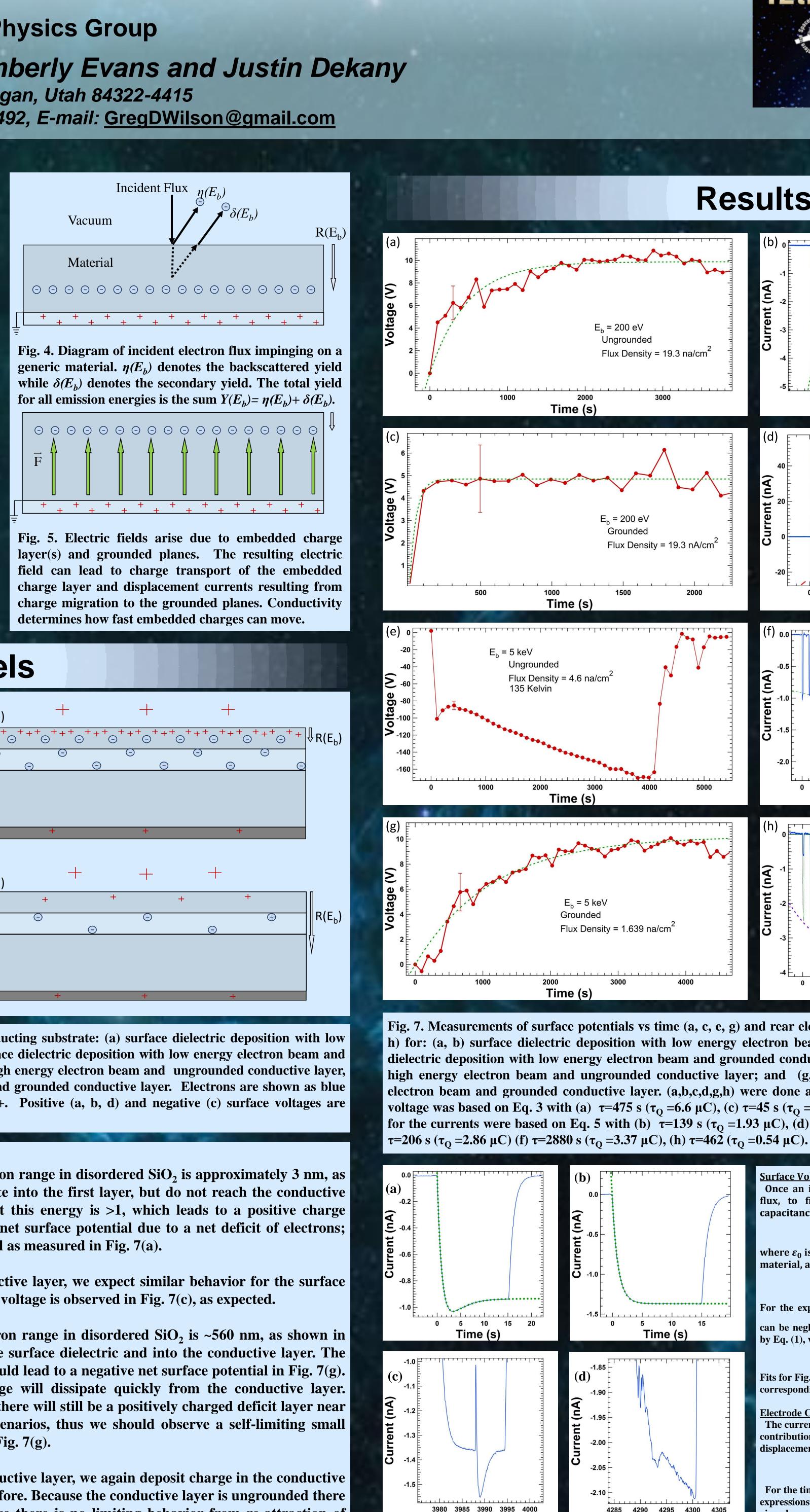


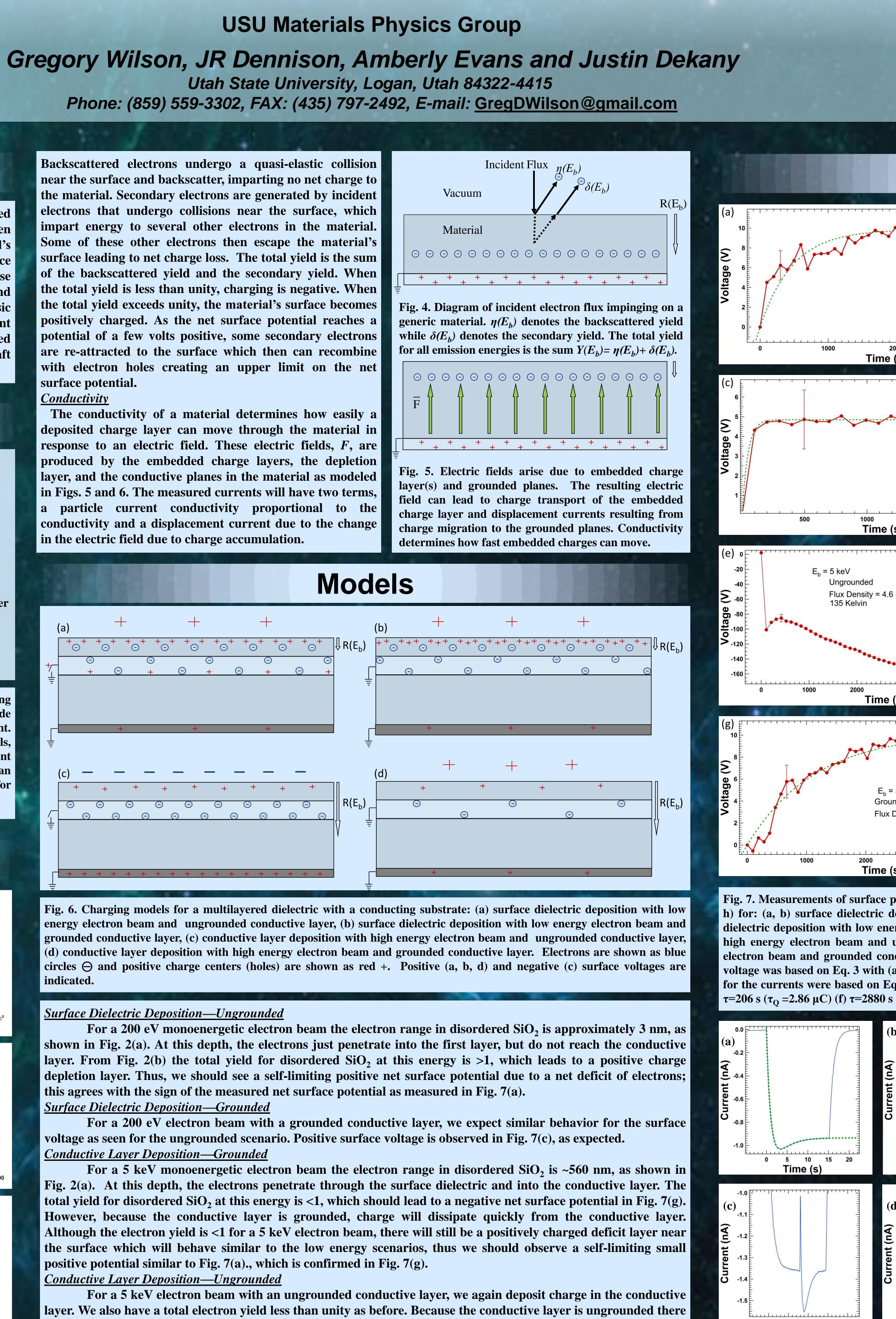
Fig. 2. (a) Electron Range  $R(E_{b})$  as a function of incident energy for Ag and for SiO<sub>2</sub>. (b) Total Electron yield as a function of incident energy for  $SiO_2$ . (c) **Resistivity as a function of temperature for SiO<sub>2</sub>.** 

# **Electron Energy Dependent Charging Effects of Multilayered Dielectric Materials**

**Backscattered electrons undergo a quasi-elastic collision** the material. Secondary electrons are generated by incident electrons that undergo collisions near the surface, which impart energy to several other electrons in the material. Some of these other electrons then escape the material's surface leading to net charge loss. The total yield is the sum of the backscattered yield and the secondary yield. When the total yield is less than unity, charging is negative. When the total yield exceeds unity, the material's surface becomes positively charged. As the net surface potential reaches a potential of a few volts positive, some secondary electrons are re-attracted to the surface which then can recombine with electron holes creating an upper limit on the net surface potential.

n Figs. 5 and 6. The measured currents will have two terms particle current conductivity proportional to th n the electric field due to charge accumulation.





will be no fast charge dissipation mechanism. Thus, because there is no limiting behavior from re-attraction of secondary electrons, we should see a high net negative potential. This is confirmed in Fig. 7(e). For this scenario, after higher negative net surface potentials were reached, breakdown and arcing was observed.

and CCD video cameras, aid with theoretical models from Alec Sim, and useful discussions with Robert Meloy and Charles Bowers of NASA Goddard Space Flight Center.

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Fig. 8. Expanded views of the rear electrode current in Fig. 7(f). ( First current pulse  $\tau_{\text{Disp}} = 0.507 \pm 0.008 \text{ s} (4.0 \pm 0.06 \text{ nC})$  and 1.4  $\pm \tau_{O} = 0.007 (11.3 \pm 0.06 \ \mu C)$ . (b) Current pulse immediately before the first observed arc  $\tau_{\Omega} = 0.966 \pm 0.001$  s (7.53  $\pm 0.007$  nC) (6 Current during first arc. (d) Current after subsequent arcing.

Time (s)

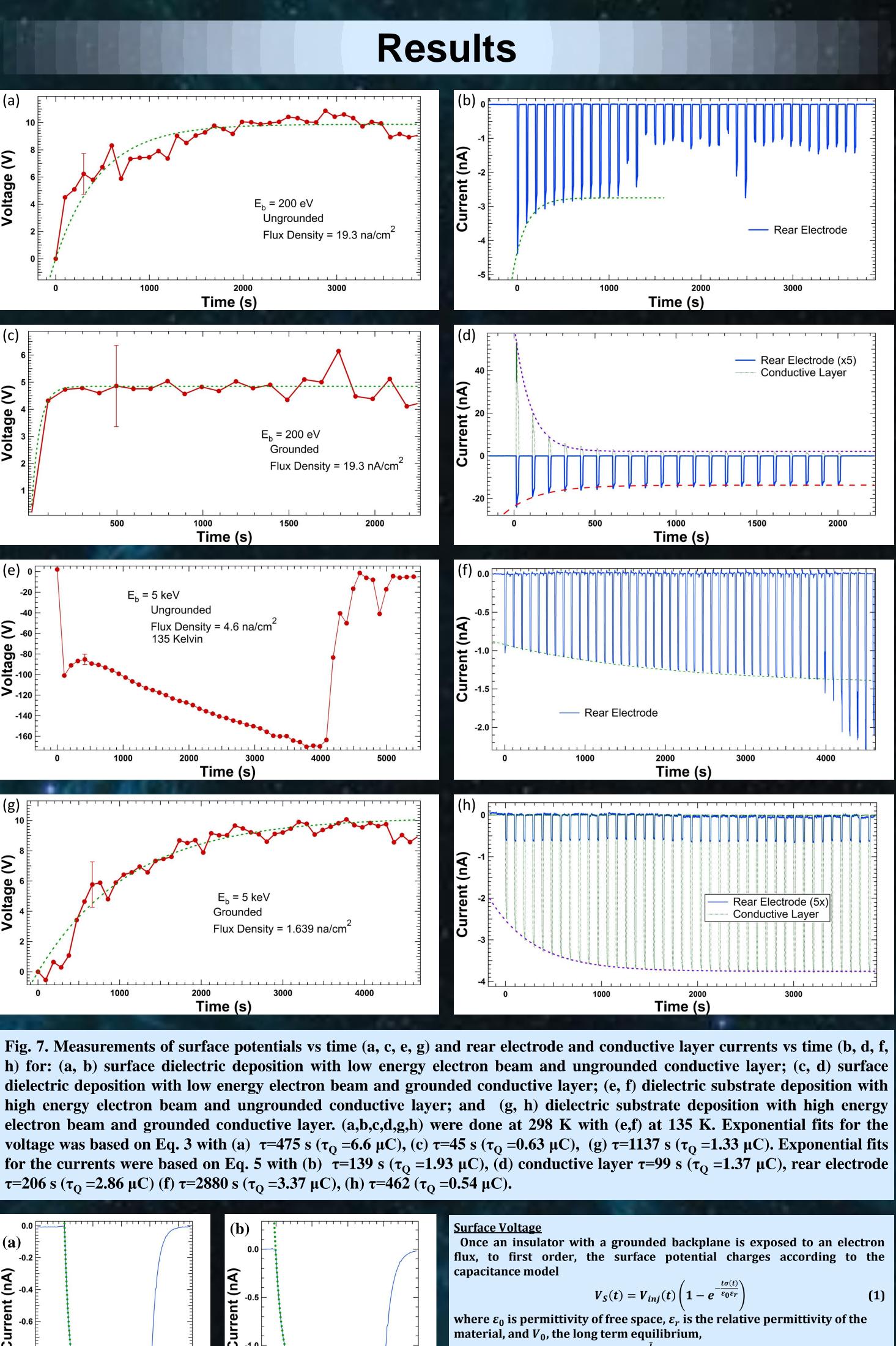
0 5 10 15

Time (s)

4285 4290 4295 4300 4305

Time (s)

# 12th Spacecraft Charging



 $V_0 = \frac{J_0}{\sigma} \left[ D - R(E_b) \right]$ 

For the experiments here,  $|\sigma(t)|_{s}$  $\int_{\mathcal{E} \circ \mathcal{E}} | \ll t$  and the exponential term in Eq. (4) can be neglected. To account for the charge dependant electron emission given by Eq. (1), we write the injection voltage as [S]

$$V_{inj}(t) = V_o(t)[1 - Y(E_b)](1 - e^{-Q(t)/\tau_Q})$$
(3)  
Fits for Fig. 7(a,c,f) are based on these exponential modes with their  
corresponding parameters reported.  
Electrode Current  
The current measured at the grounded rear electrode includes two  
contributions, the free charge transport current density,  $J_c$ , and the charge  
displacement current density,  $J_{displacement}$ .  
 $J_{elec}(t) = J_{elec}^c(t) + J_{elect}^{displacement} = \sigma(t)F(t) + \epsilon_o\epsilon_r \frac{\partial F(t)}{\partial t}$ (4)  
For the time independent conductivity estimated above and for general values

expressions for the parallel plate geometry, it can be shown that this current given by

 $J(t) = \bar{J}_0(t) [1 - Y(E_b)] (1 - e^{-Q(t)/\tau_Q})$ 

Fits based on these models, with the displacement current neglected due to le time frames, are shown in Fig 7(a,d,f,h) with their respected values report Figure 8(a,b) also have fits based on these models but (a) also includes an exponential for the displacement current. After several beam pulses the displacement current dies out as shown in Fig. 8(b).