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## Dependence of Electrostatic Field Strength on Voltage Ramp Rates for Spacecraft Materials

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*Utah State University*

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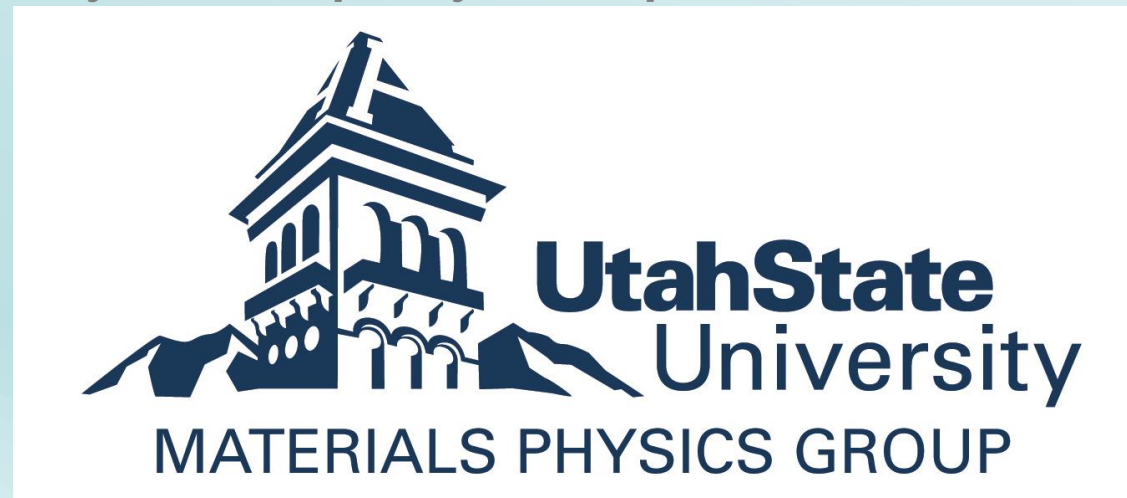
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# *Dependence of Electrostatic Field Strength on Voltage Ramp Rates for Spacecraft Materials*

**Krysta Moser, Allen Andersen, and JR Dennison**

*Materials Physics Group, Physics Department, Utah State University*



2015 USU Physics Colloquium

# Acknowledgements

## Support & Collaborations

*USU Blood Fellowship*  
*NSTRF Fellowship*



Alec Sim



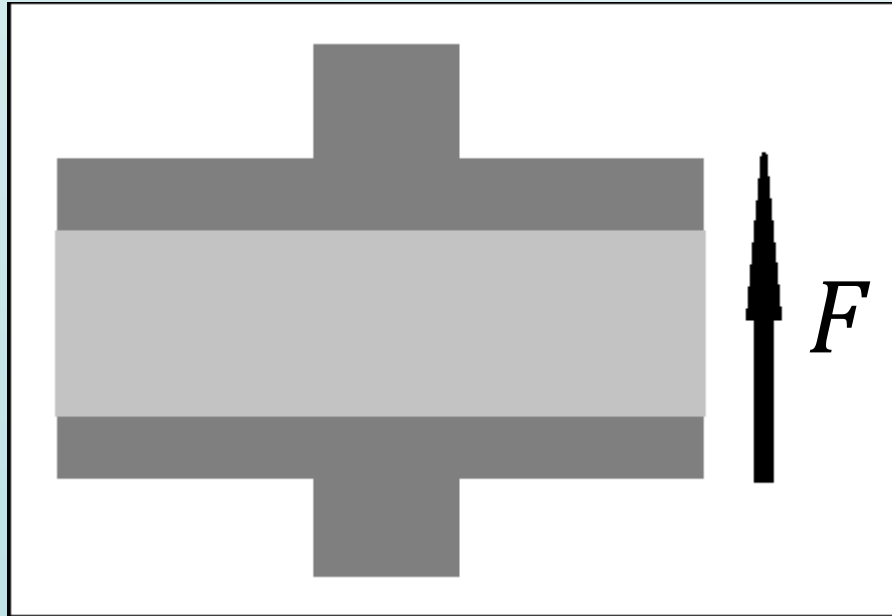
Charles Sim



Allen Andersen



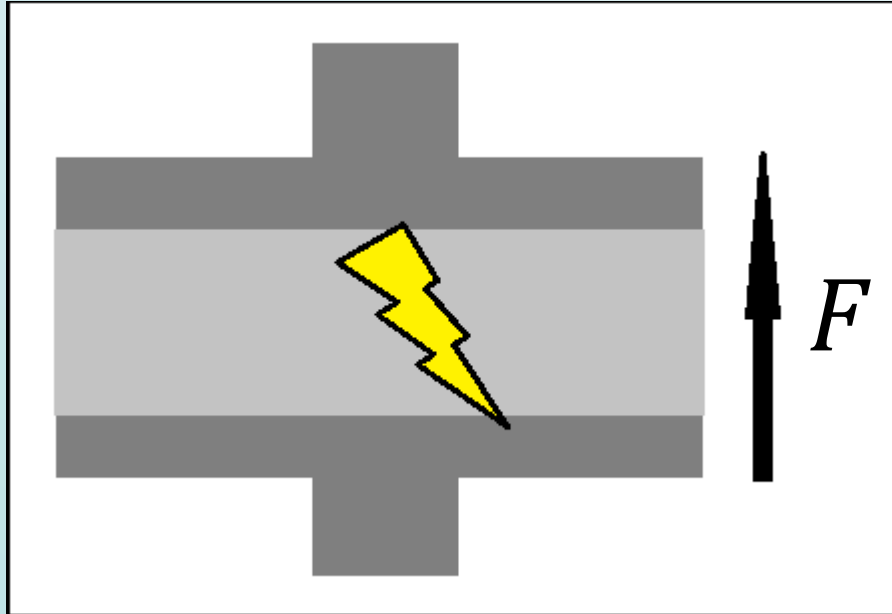
USU's Materials Physics Group



Consider a simple parallel plate capacitor.

- At low fields current flow is restricted.

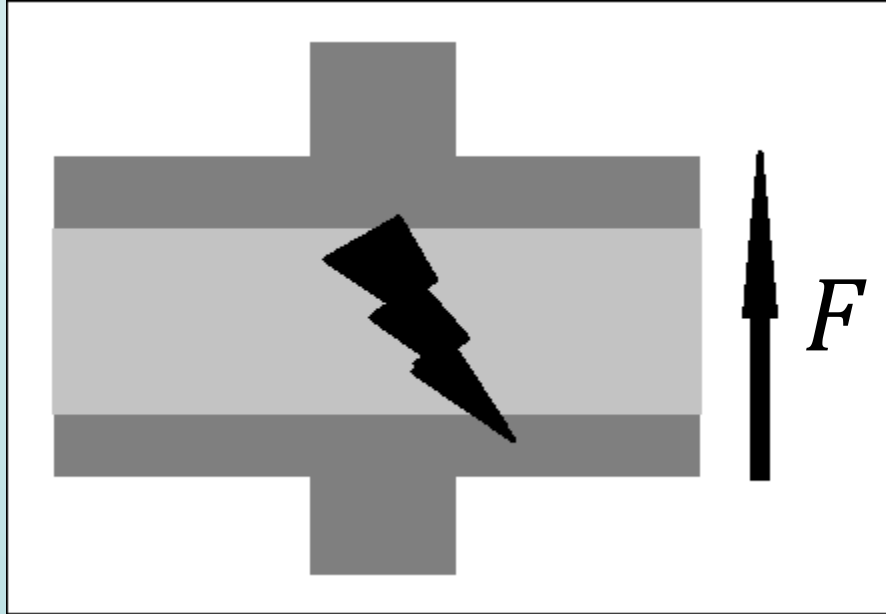
# Introduction—What is ESD?



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- At high enough fields or after long times the insulator can breakdown.
- Large currents can flow.

# Introduction—What is ESD?



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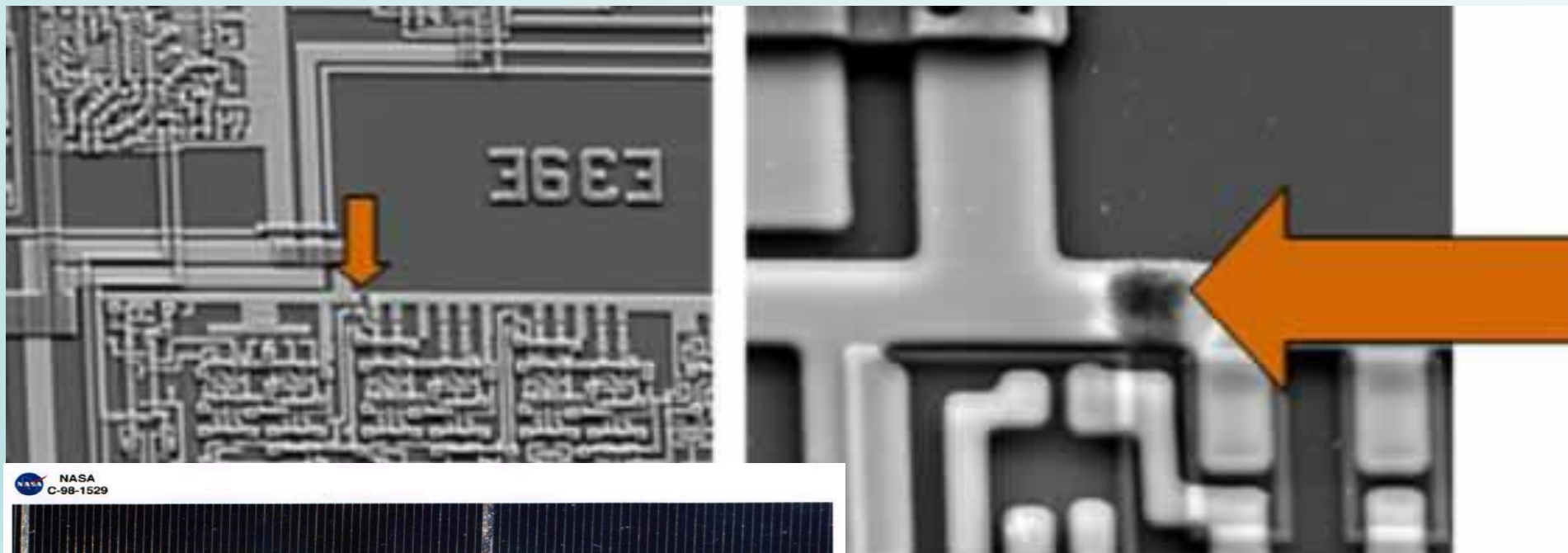
- At low fields current flow is restricted.
- At high enough fields or after long times the insulator can breakdown.
- Large currents can flow.
- Electrostatic discharge (ESD) is a **permanent, catastrophic failure** of a dielectric material.
- What was an insulator is now essentially a conductor in the system.

# Why should we care about ESD?

ESD is the primary cause of failures attributed to spacecraft interactions with the plasma space environment.



# Why should we care about ESD?



Smaller  
devices=bigger  
problems



Solar panel damaged by localized charging event

National Aeronautics and Space Administration  
Lewis Research Center

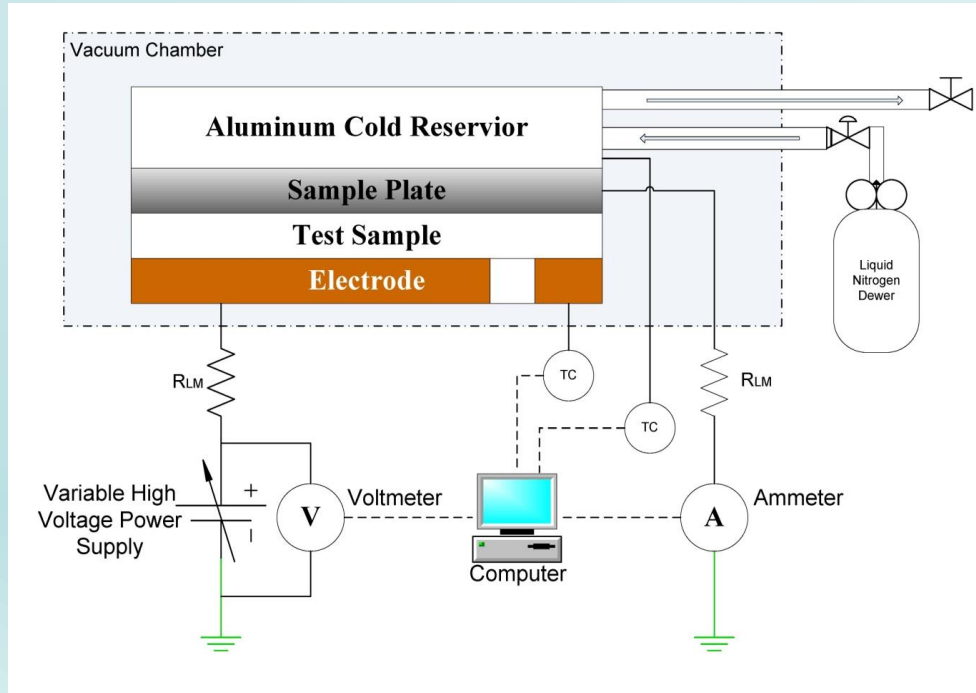
Electrostatic field strength is affected  
by many factors

Important to understand how  $F_{ESD}$   
varies under different conditions

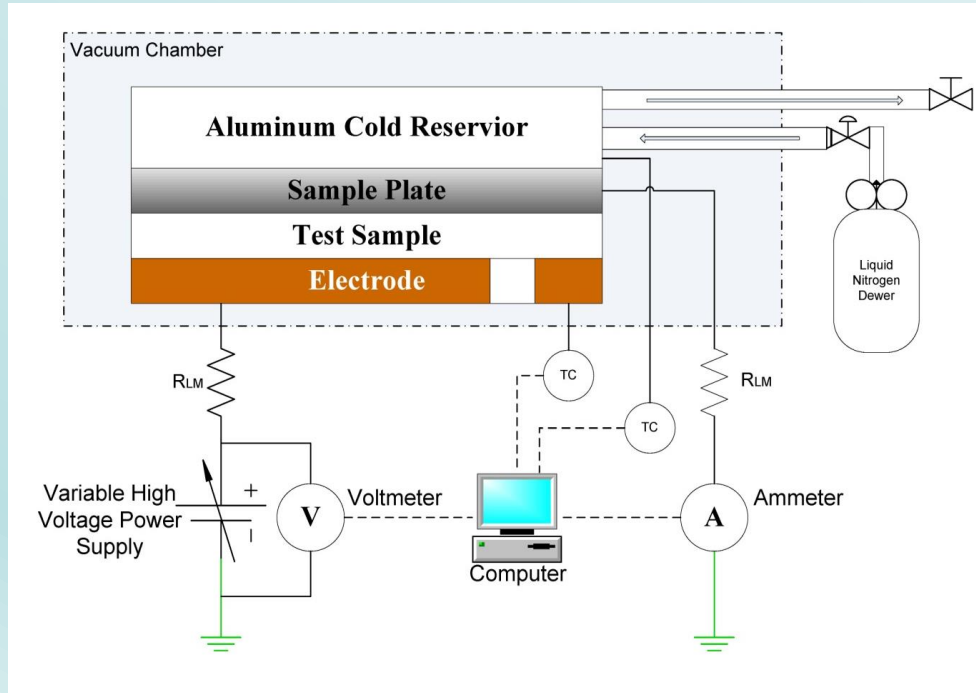
Past work in USU's MPG has looked at  $F_{ESD}$   
dependence on temperature, voltage ramp rate,  
duration of applied electric field, etc



# ESD Vacuum Chamber



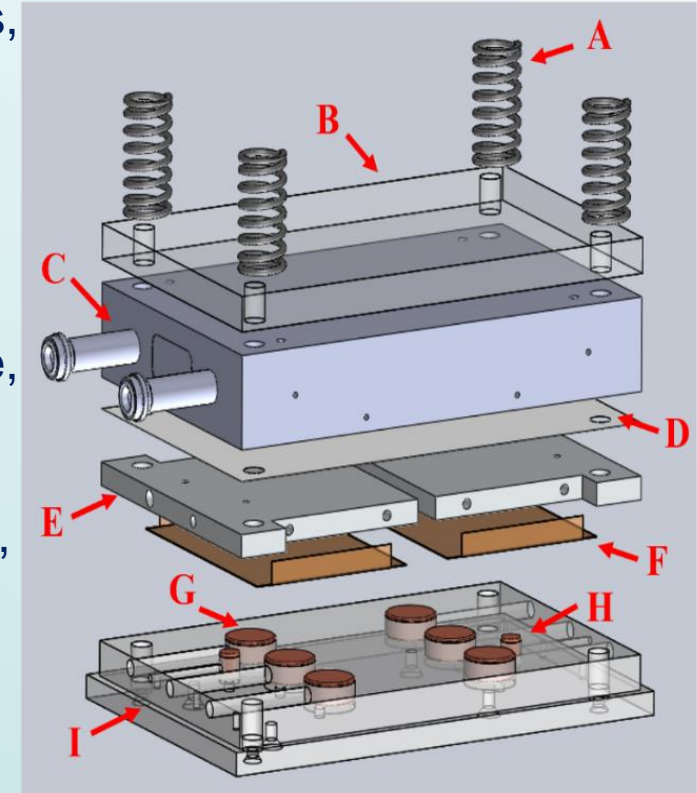
- Simple parallel plate capacitor
- Vacuum  $\sim 10^{-6}$  torr
- Applies up to 30 kV
- 6 electrode carousel



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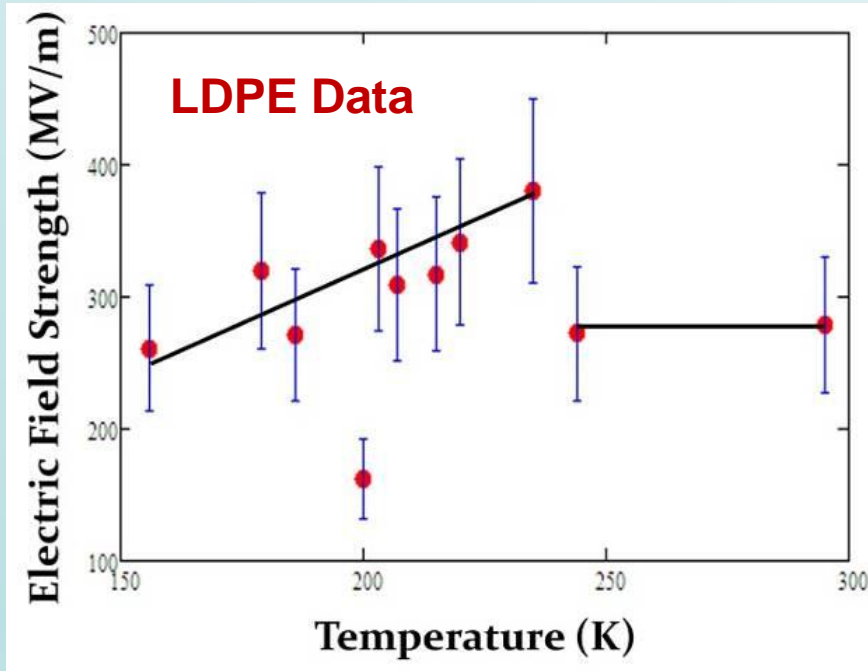
## ESD Test Assembly:

- (A) Adjustable pressure springs,
- (B) Insulating layer
- (C) Cryogen reservoir,
- (D) Thermally conductive, electrically isolating layer,
- (E) Sample and mounting plate,
- (F) Sample
- (G) HV Cu electrode
- (H) Cu thermocouple electrode,
- (I) Insulating base.

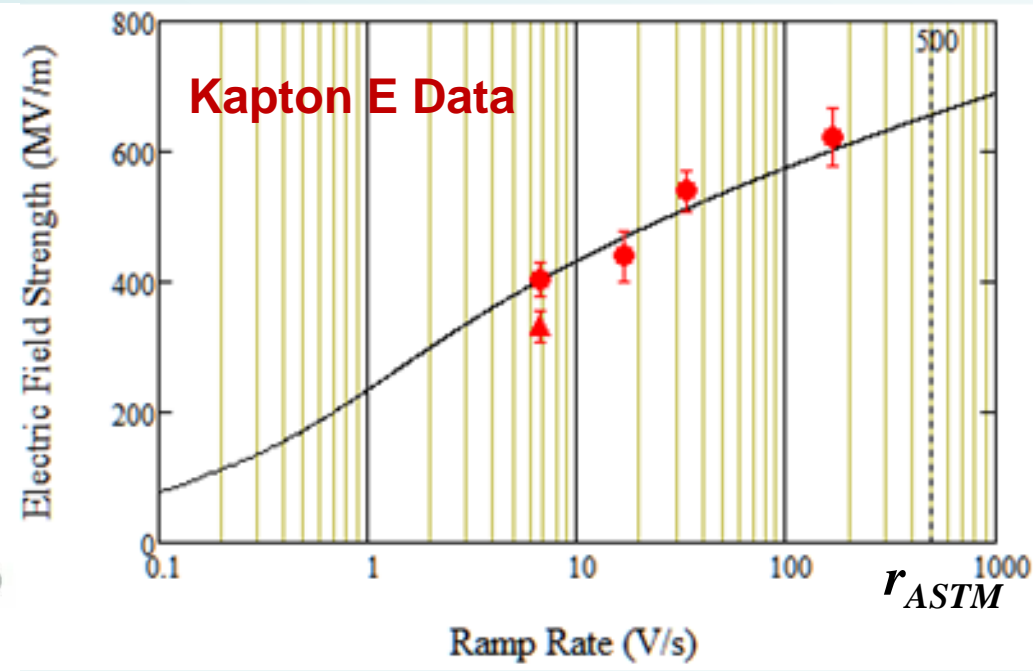


# Breakdown Test Dependence on T and dV/dt

## $F_{ESD}$ Temperature Dependence



## $F_{ESD}$ Ramp Rate Dependence



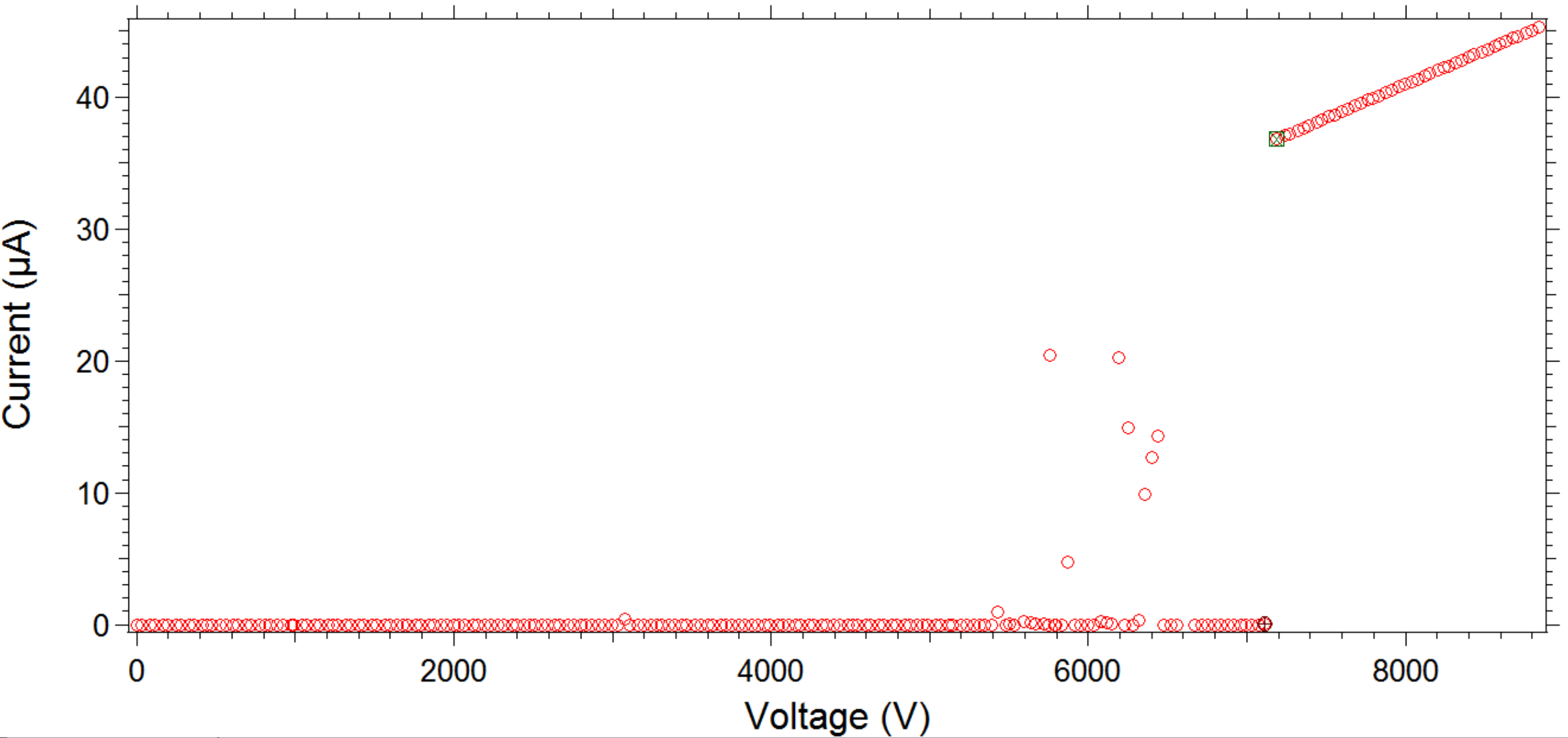
$F_{ESD}$  depends significantly on both temperature and ramp rate.

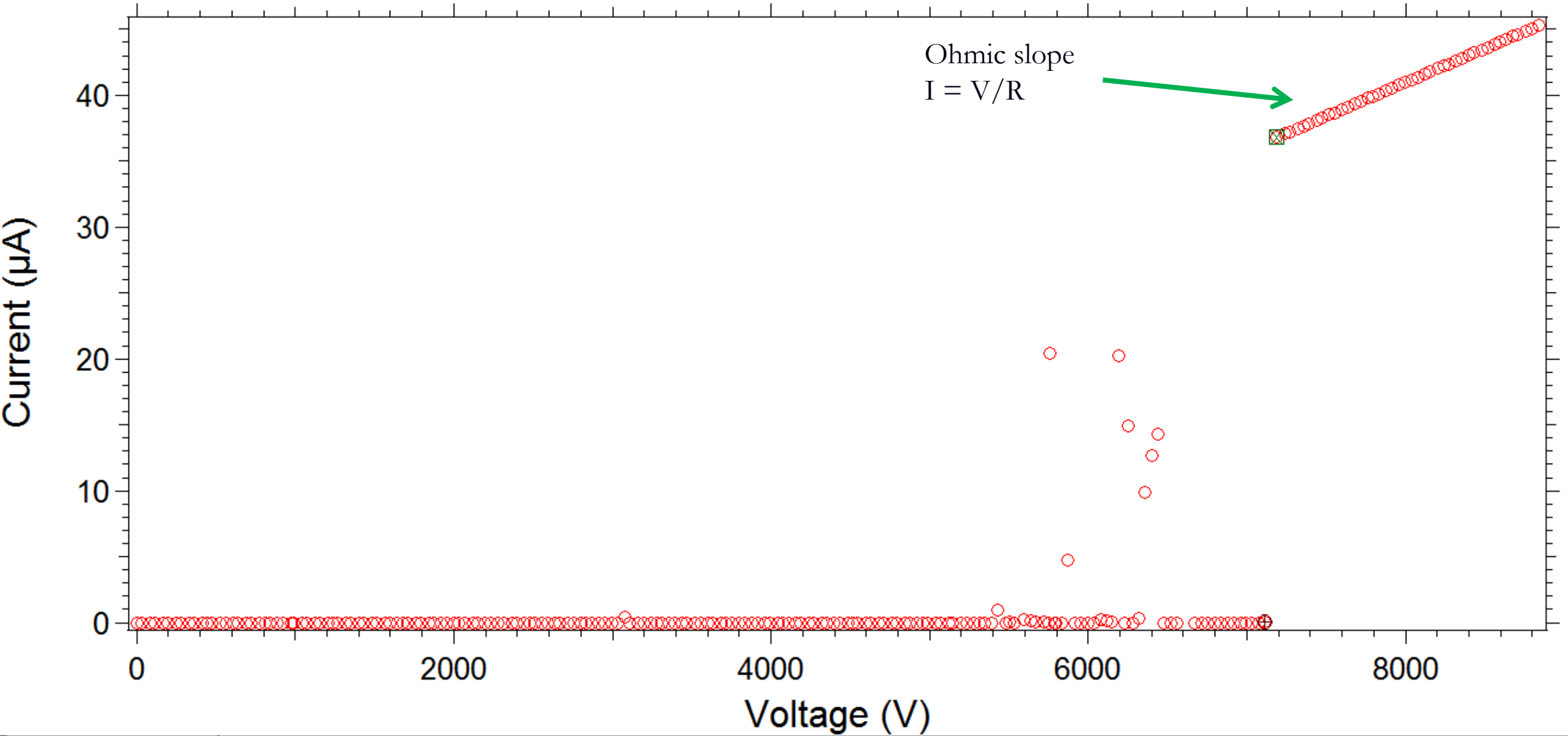
- ❑ ASTM D3755 standard tests recommend a 500 V/s ramp rate until breakdown.
- ❑ However these tests are not very repeatable and tend to overestimate breakdown strengths for slower ramp rates.
- ❑ Slow (even VERY SLOW) ramp rate better model real charging applications.

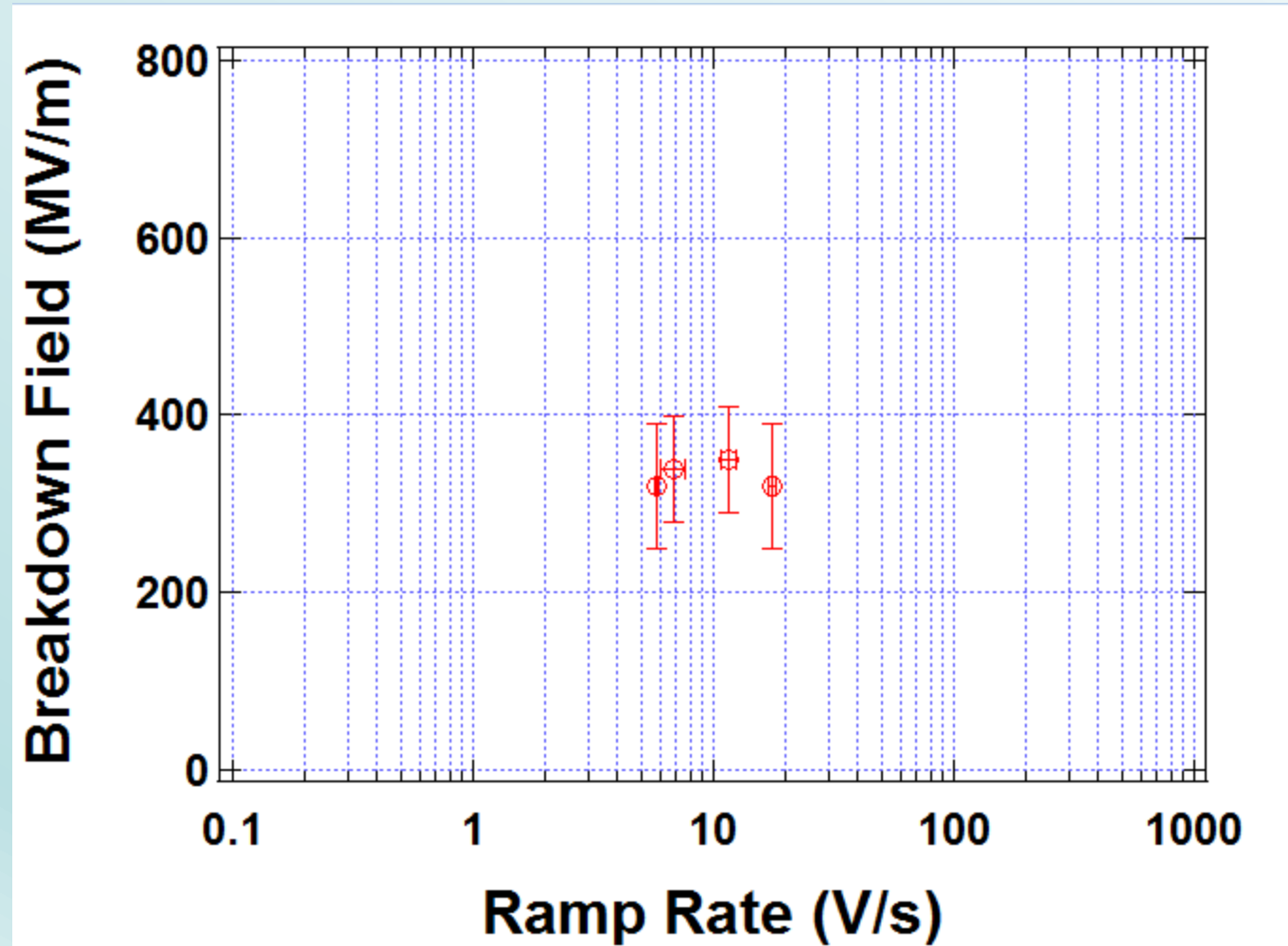
**If you use the recommended ramp rate of  
500 V/s to test spacecraft charging**

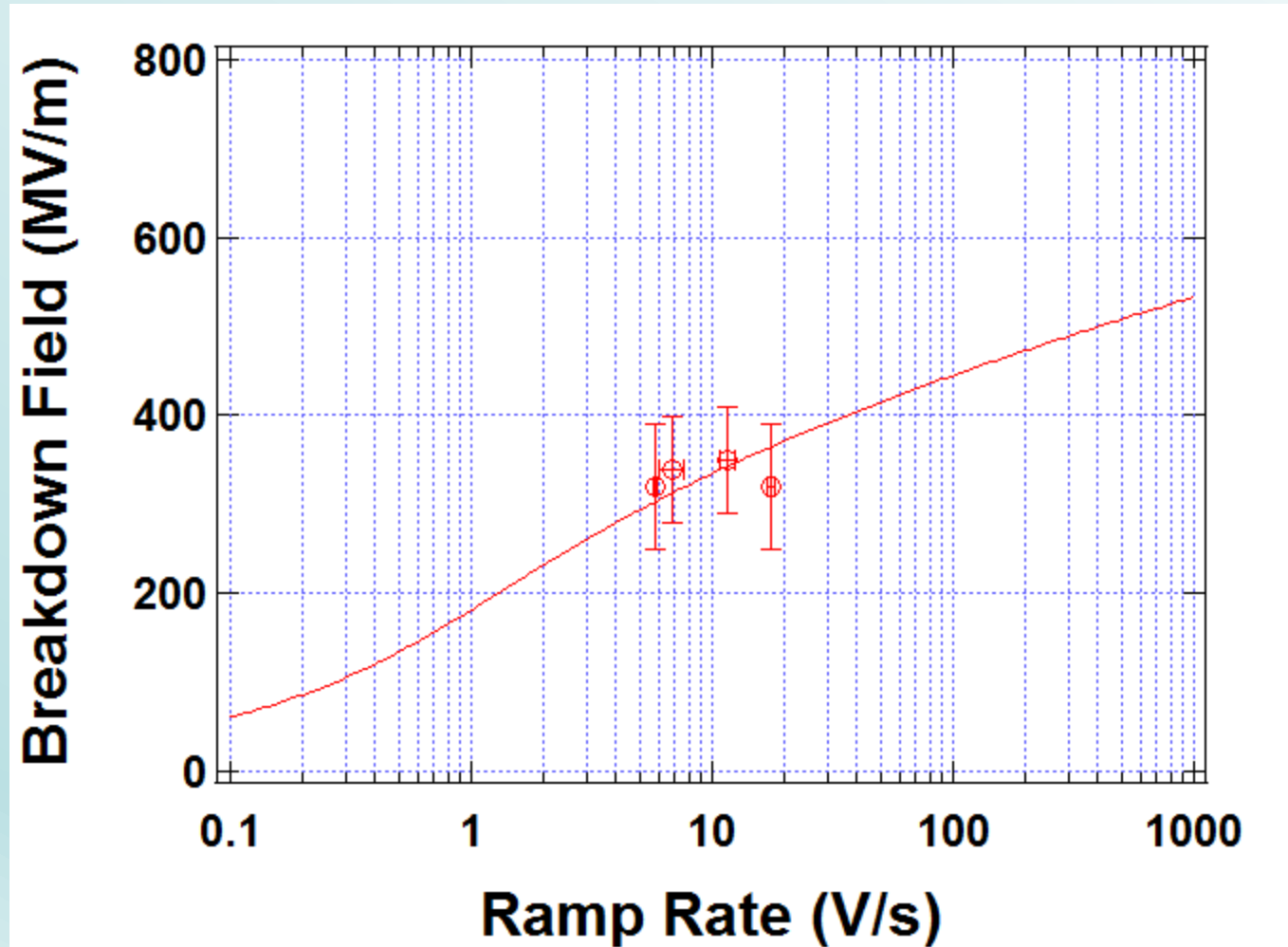


**You're gonna have a bad time**



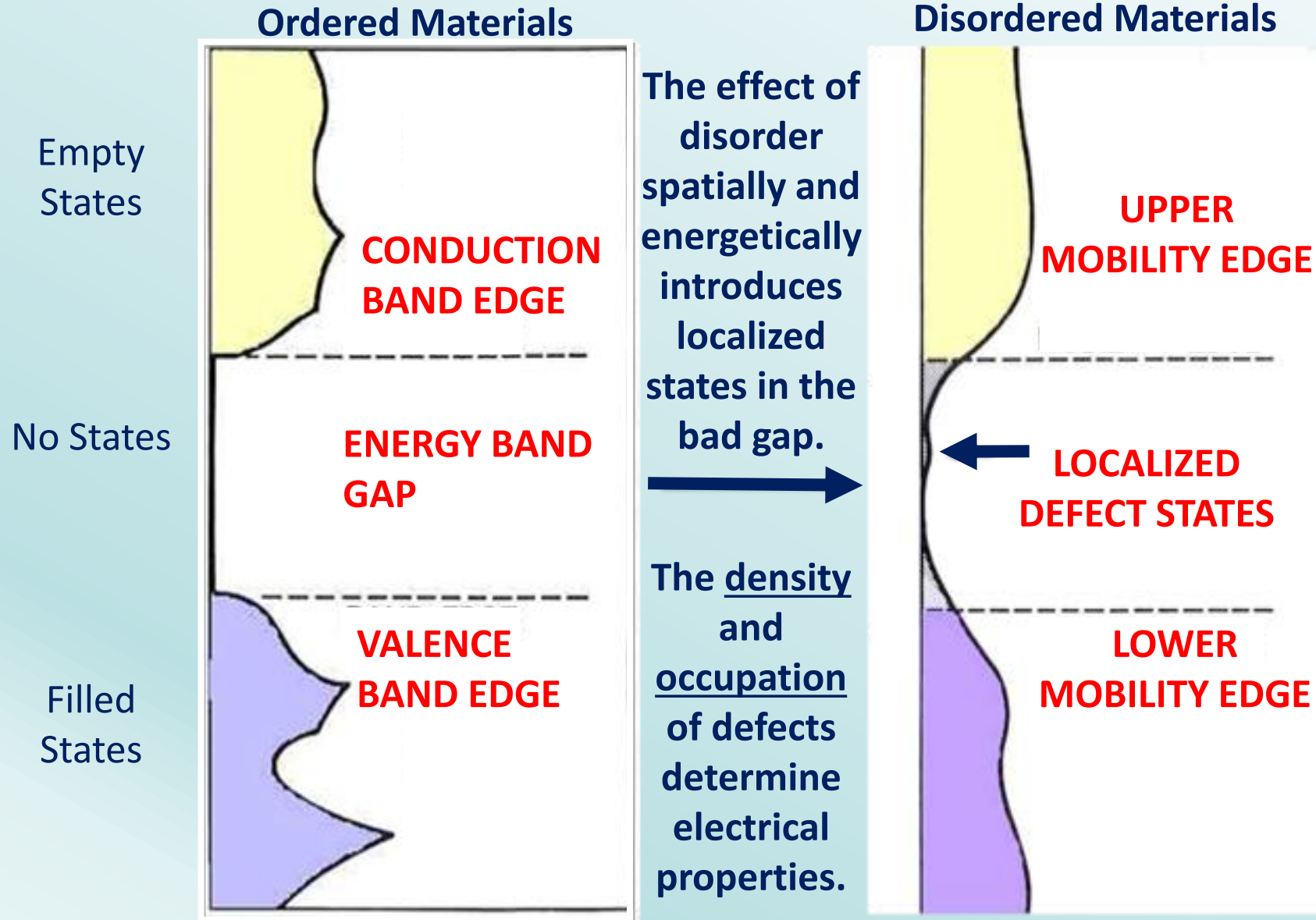


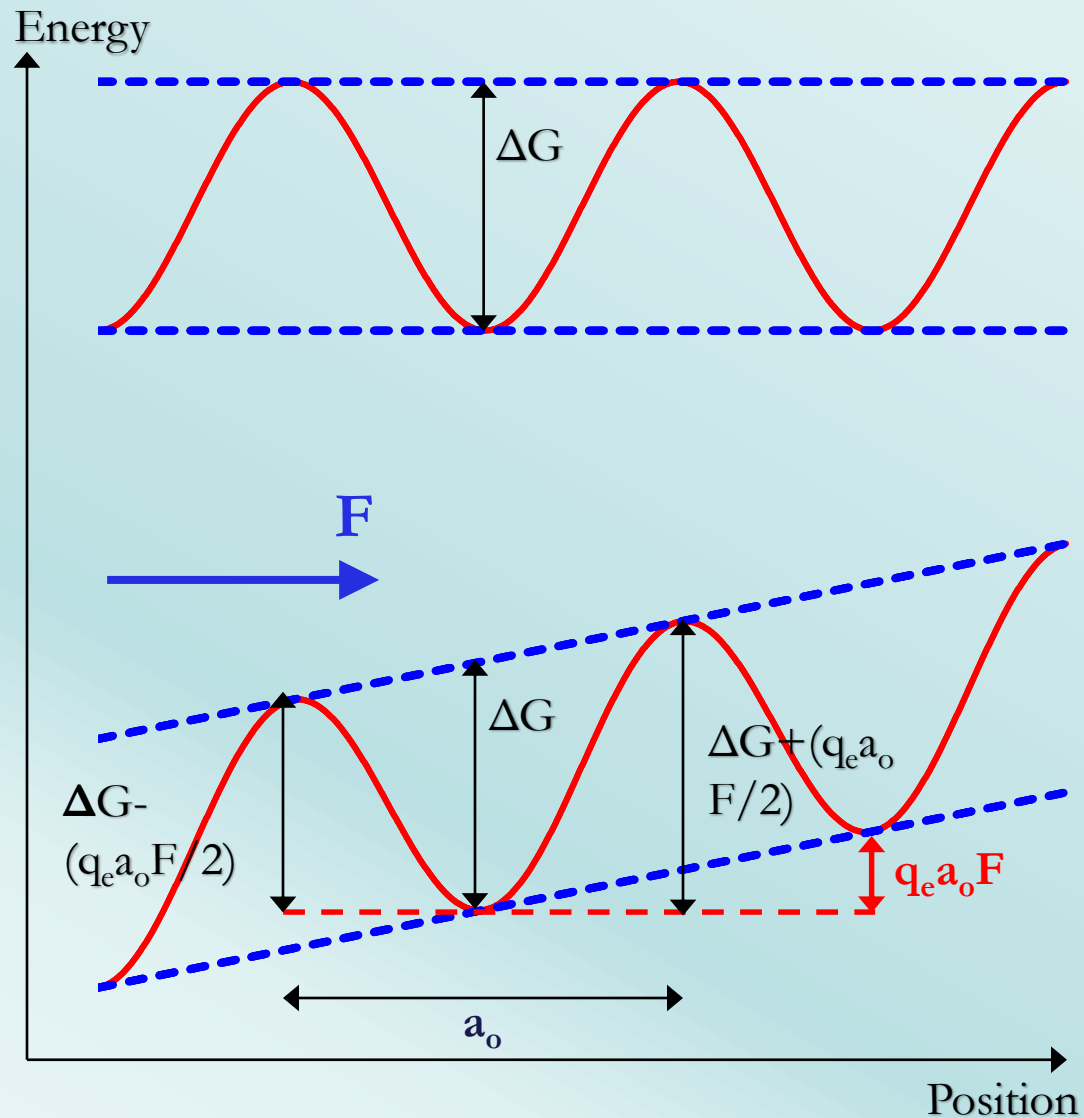




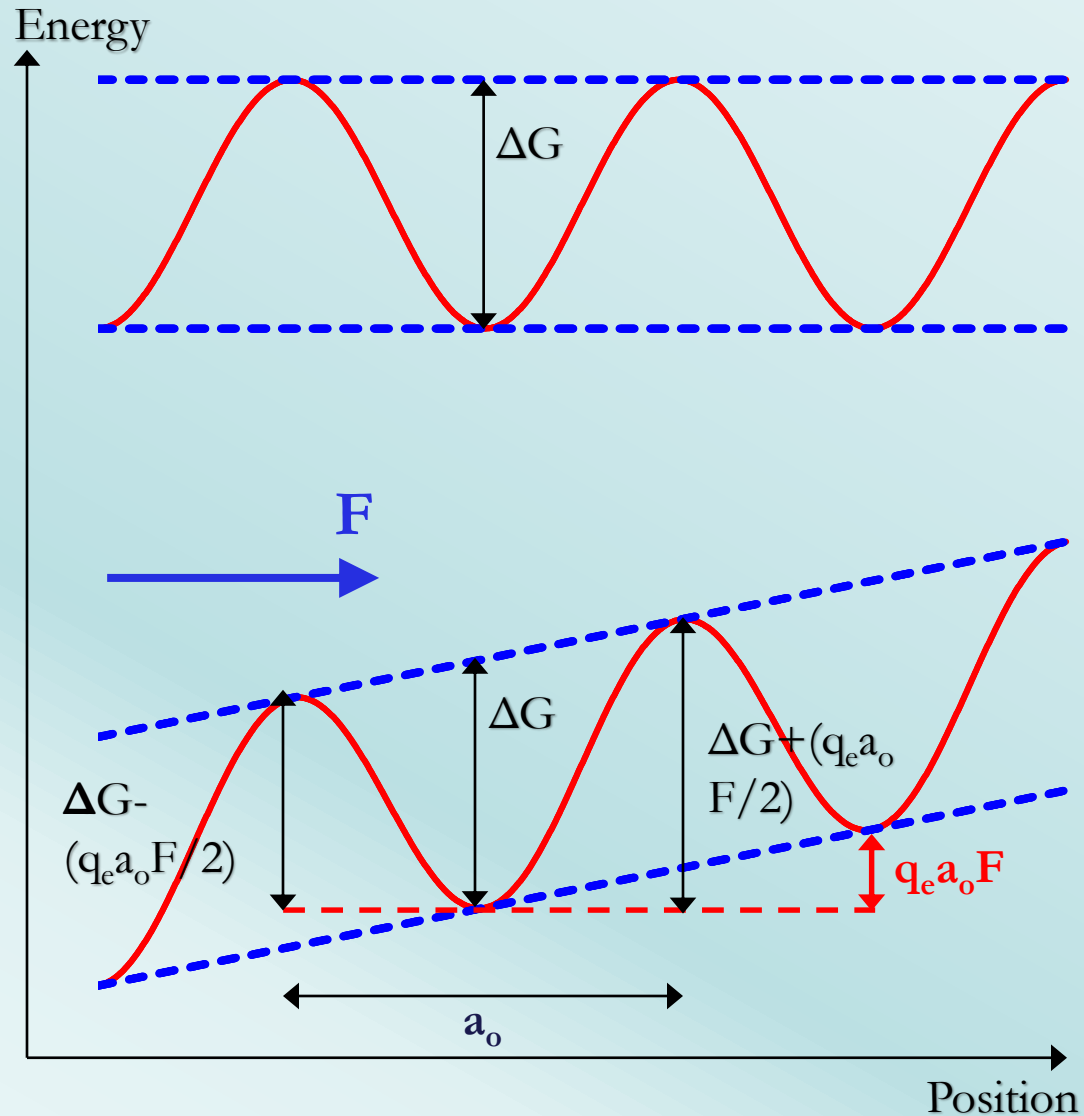
$$F_{ESD}(r) \approx F_{ESD}(r_0) \sqrt{1.1346 \ln(r + \sqrt{1 + r^2})}$$







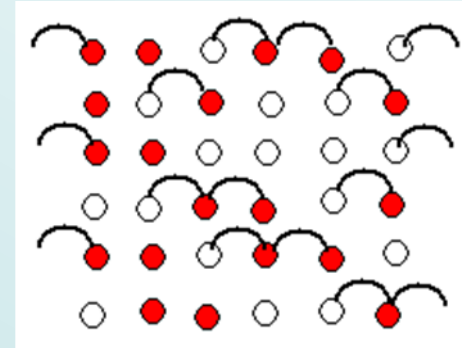
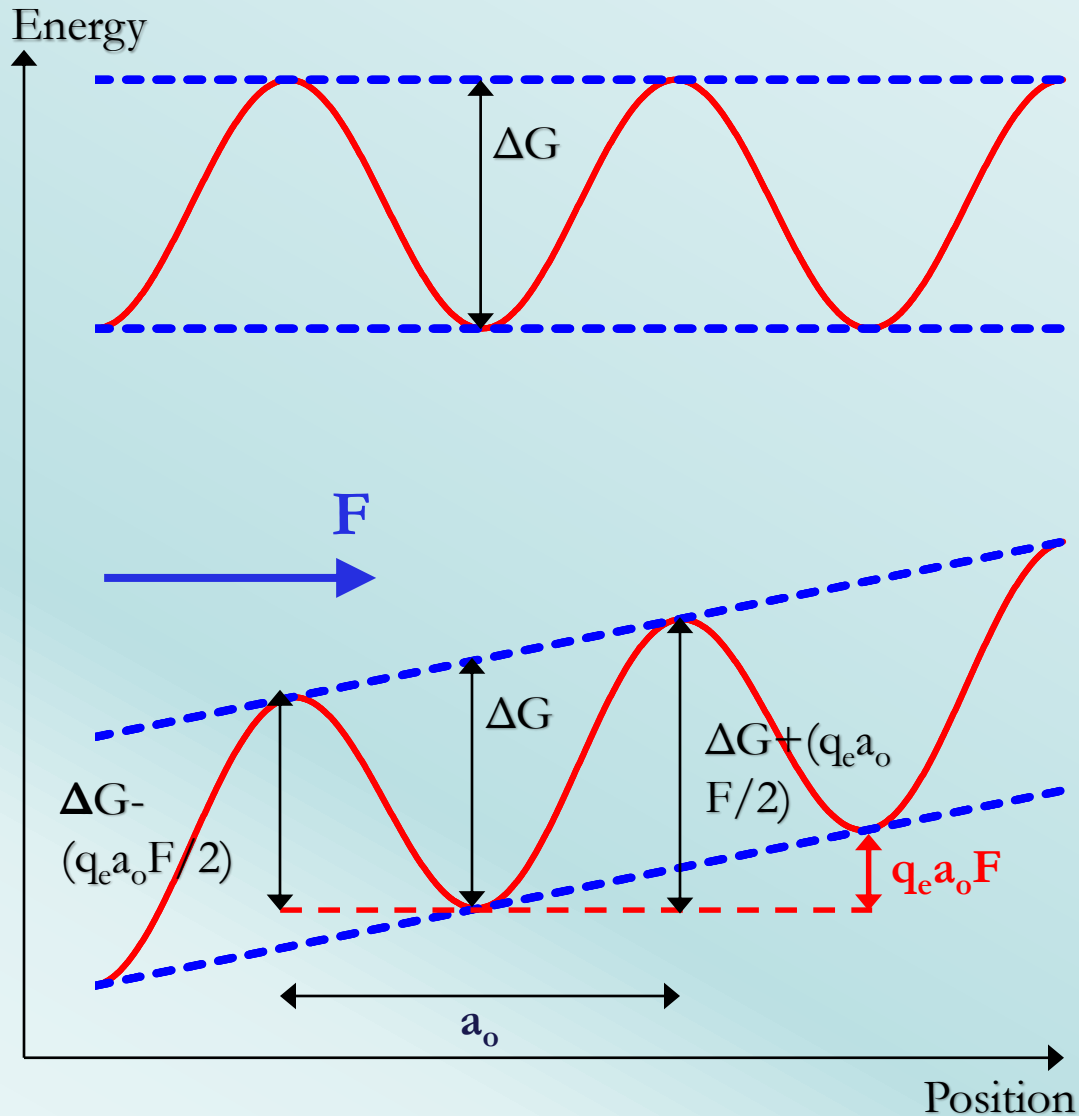
Under an applied electric field, charge can 'hop' between defect sites. The probability of a transition in a given time step depends upon temperature, well depth, and applied field.



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$$F = qeE$$

$$W = F * a = qeaF$$

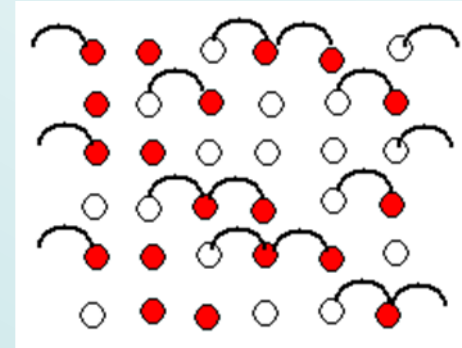
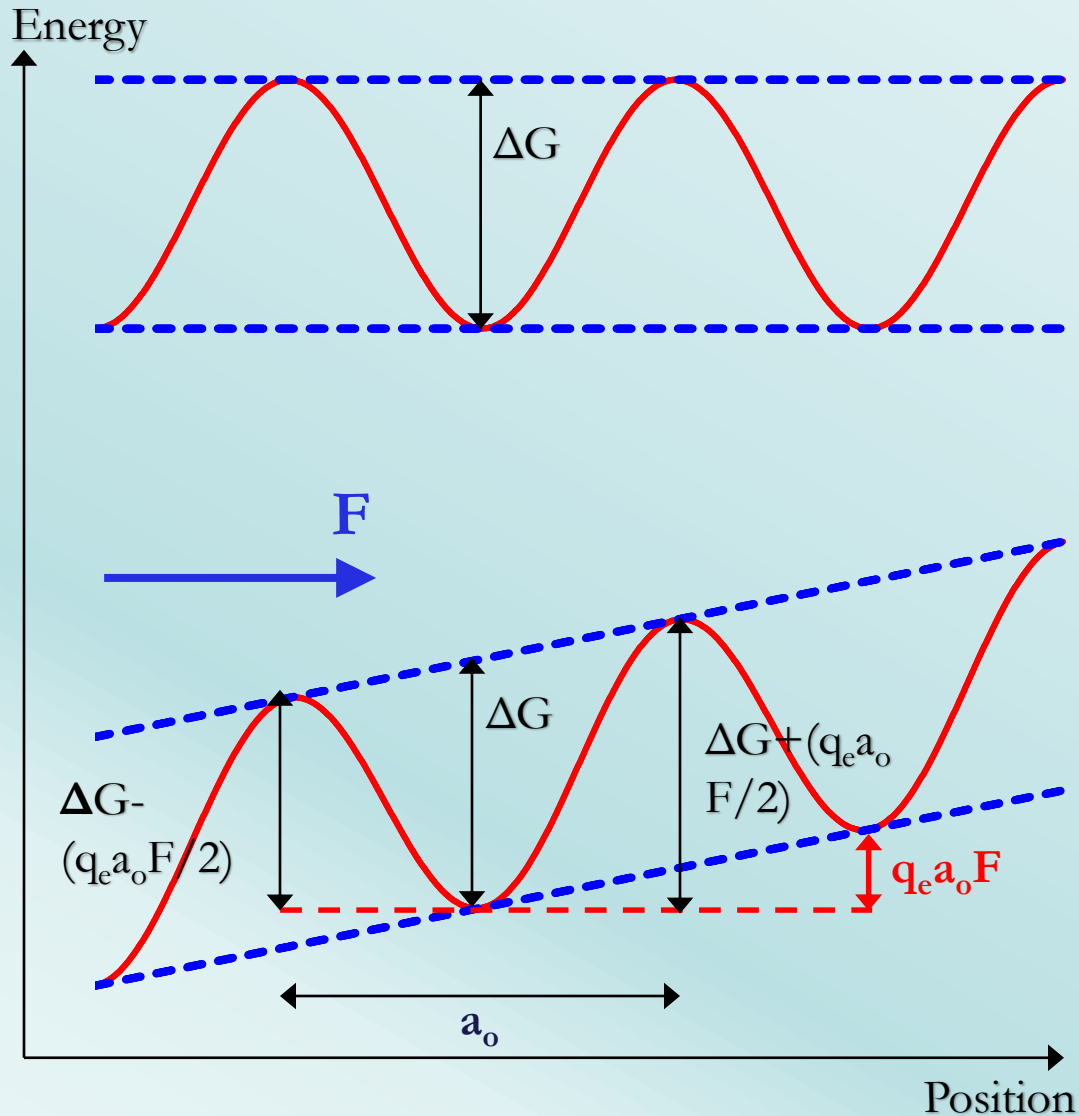


Assumes all depths and hopping distances are the same

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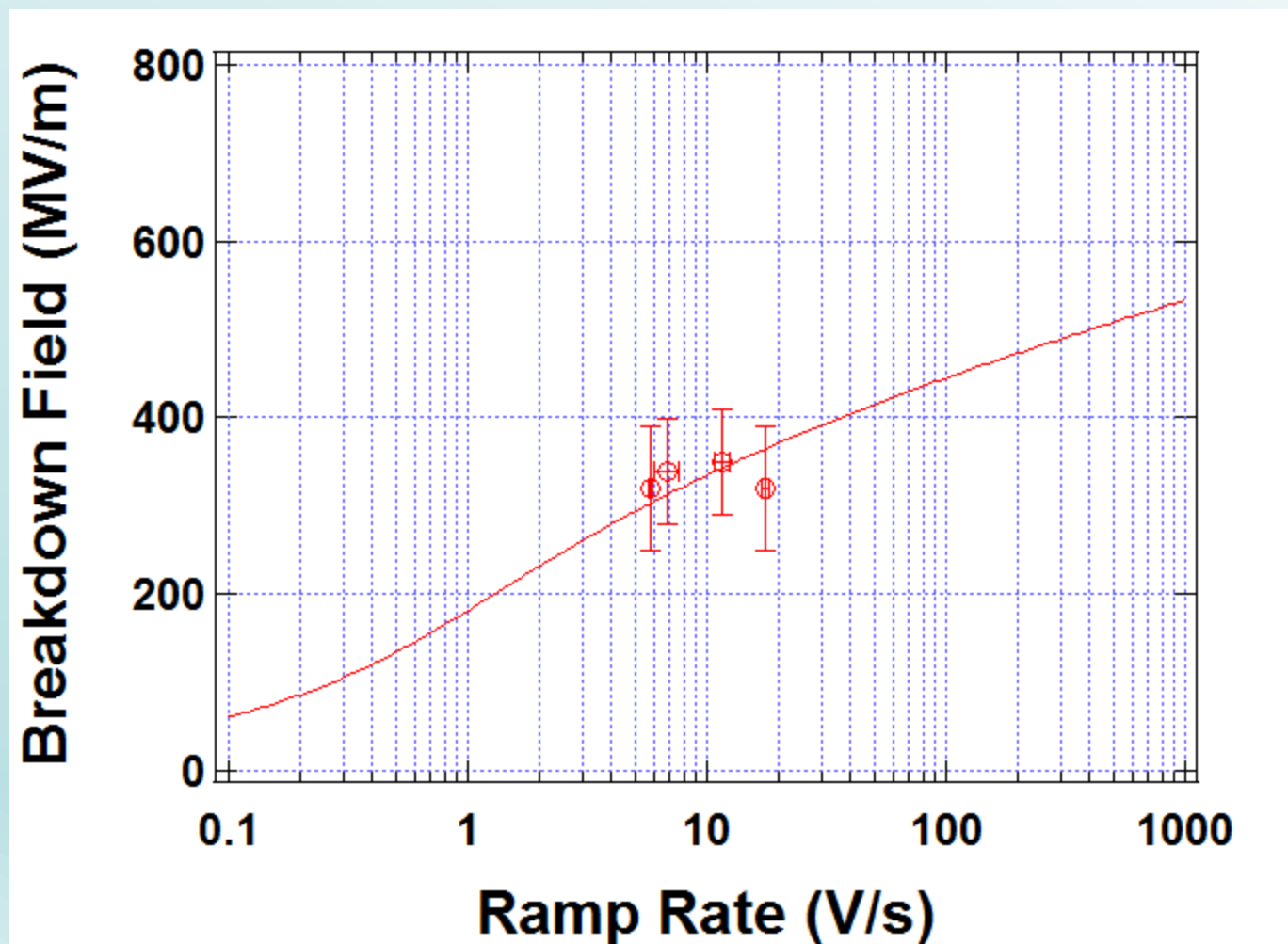
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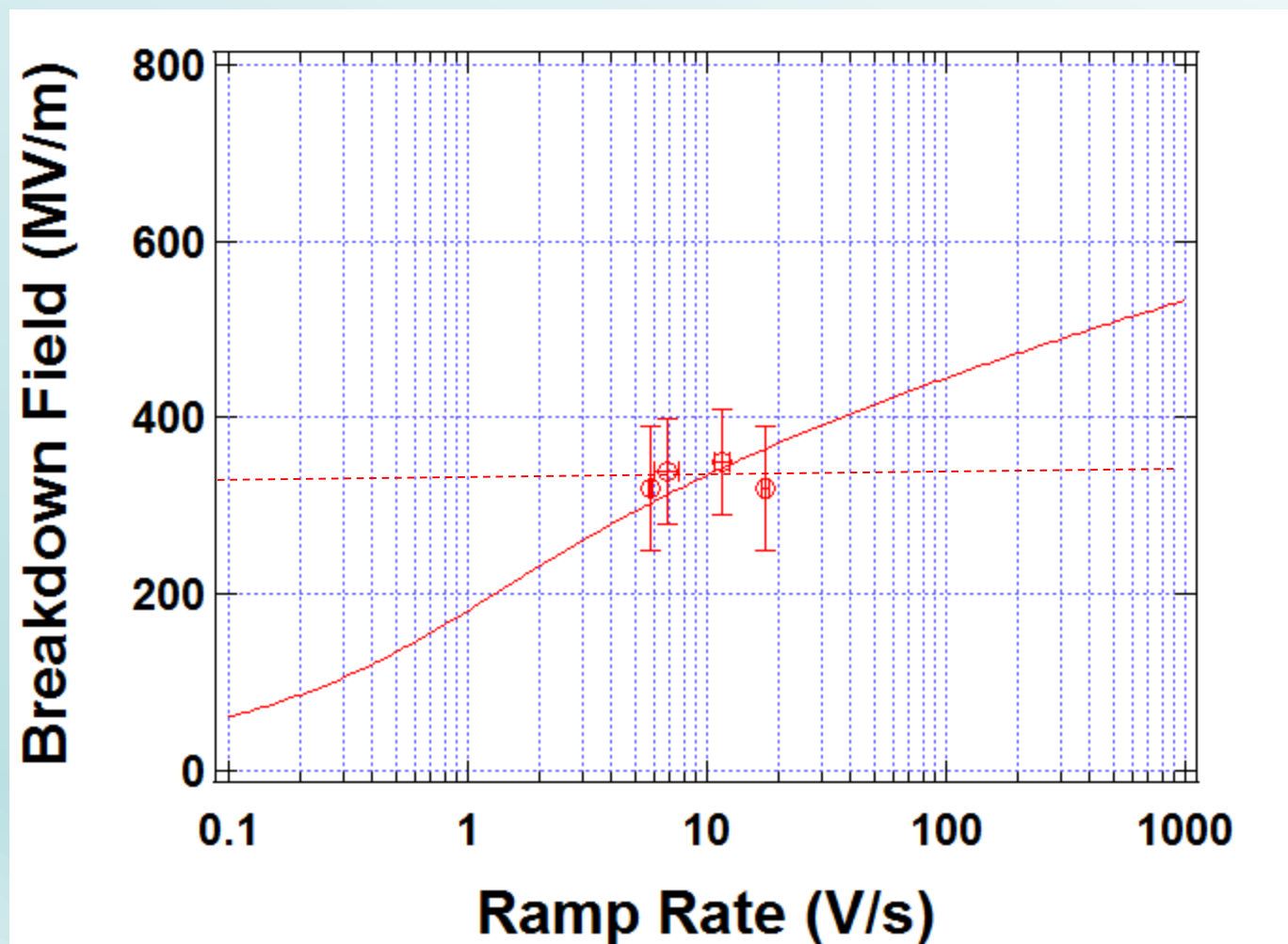
$$W = F * a = qeaF$$

$$P_{def}(F, T, \Delta t) = \left( \frac{2k_B T}{h/\Delta t} \right) \exp \left[ \frac{-\Delta G_{def}}{k_B T} \right] \sinh \left[ \frac{\epsilon_0 \epsilon_r F^2}{2k_B T N_{def}} \right]$$

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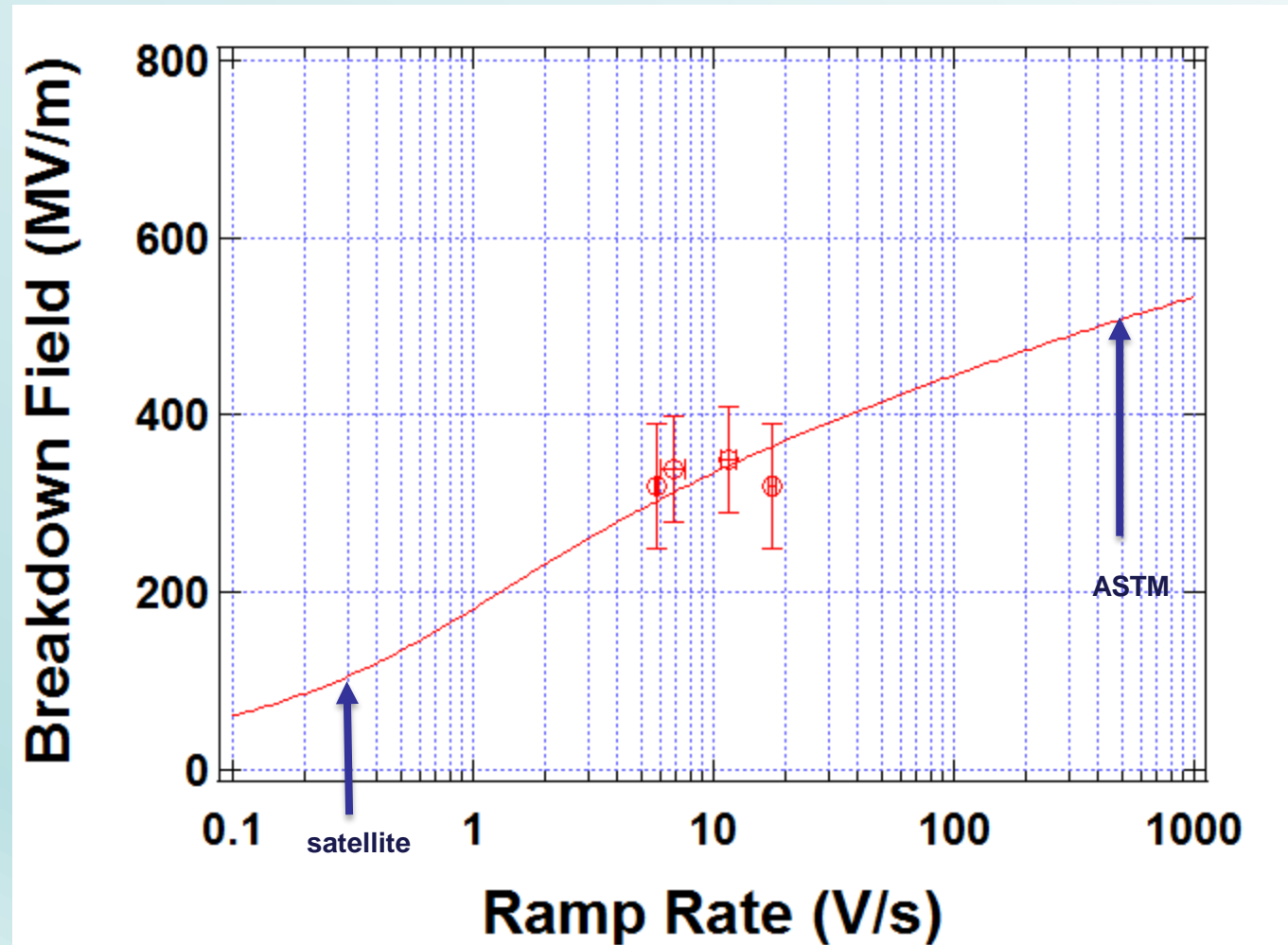


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# Future Work



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## Future Work

- **Look into ramp rates on extreme ends of graph**
- **Expand ramp rate tests already done on Kapton/LDPE**
- **Begin testing temperature dependence of all three**

# References

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