VIIRS On-orbit Optical Anomaly

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> Presented by Jack Xiong NASA/GSFC 618.0

Folks who did the work

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Introduction

The NPOESS Preparatory Project (NPP) spacecraft was launched on Oct. 28, 2011, and began the commissioning phase of several of its instruments shortly thereafter. One of these instruments, VIIRS, was found to exhibit a gradual but persistent decrease in the optical throughput of several bands, with the near-infrared bands being more affected than those in the visible. This paper will discuss the sequence of events and activities which identified the cause of the optical degradation and the current and expected future state of the optics.

A year ago today

On Aug. 28, 2011, NPP was placed in a shipping container and loaded on a transport truck at Ball Aerospace & Technologies Corp.

2 months late

On October 28, 2011, NPP was launched from Vandenberg Air Force Base (VAFB), CA

http://www.launchphotography.com/NPP.html

Ben Cooper / LaunchPhotography.com

Visible/Infrared Imager Radiometer Suite (VIIRS)





22 bands: 0.4 to 12.5 µm

VIIRS OMM and SDSM Solar Calibration Components



Source/Credits: JPSS RTN VIIRS F2F Deep Dive –Data subject to RTN Proprietary and Sensitive Data Restrictions

Two things were noticed

VIIRS NIR detector response degrades much faster than expected.

SDSM detector response also degrades at NIR wavelengths.

Led to a series of reviews, investigations, tests, data analyses,





Event	Description	Start-End	Start Data Tima	End Data Time	Start-End
#	Description	DOY			Orbit
1	NPP Launch	301-301	10/28/2011_09:48z	10/28/2011_09:48z	0-0
2	VIIRS Turned On	312-312	11/08/2011_05:44z	11/08/2011_05:44z	154-154
3	VIIRS Door Open	325-325	11/21/2011_16:03z	11/21/2011_16:03z	344-344
4	VIIRS 1394 Anomaly	329-330	11/25/2011_16:36z	11/26/2011_19:38z	401-417
5	Svalbard communication Loss (Cable cut)	331-336	<u>11/28/2011_11:00z</u>	12/02/2011_24:00z	441-505
6	RTA stowed for 3 days Sensor Test 1 (ST-1)	343-346	12/09/2011_22:47z	12/12/2011_19:33z	604-644
7	RTA rotating at night only, Sensor Test 2 (ST-2)	349-353	12/15/2011_16:32z	12/19/2011_19:37z	685-744
8	RTA Stowed for one week Sensor Test 3 (ST-3)	355-362	12/21/2011_17:14z	12/28/2011_14:55z	771-869
9	RTA Stowed for 4 days Sensor Test 4 (ST-4)	363-002	12/29/2011_14:36z	01/02/2012_20:18z	883-943

Event-5: This is just a data gap due to communicatio issue between Svalbard and IDPS. All RDRs are being processed and SDR gap will covered.

Anomaly Overview

- VIIRS VisNIR degradation first reported 12/7/2011
 - Degradation rate increased sharply after nadir door opening on 11/21/11
 - Degradation essentially stops when stowed; sensitive to UV exposure
 - DNB response consistent with observed VisNIR degradation
- SDSM has a different optics path that shows similar spectral degradation
 - Degradation continues during stow period; insensitive to UV exposure
- VIIRS solar diffuser does not show this NIR spectral degradation
 - Showing nominal expected degradation in shorter wavelengths

VNIR Response vs Orbit Time Trends

(Aerospace Data Plot)

Degradation is uniform across detectors within a band



VNIR Response vs UV Exposure Time



The degradation was correlated with UV exposure.

SDSM Normalized Response (NASA Data Plot)

672, 870 & 926 nm Detectors show largest degradation

- SDSM NIR degradation appears to be unchanged through Stow period, contrary to VIIRS VNIR trend.
- SDSM exposed to light ~6 minutes per orbit during Solar Cal period and has been constant since Nov. 8th, except for 3 Stow periods:
- All other times during orbit, SDSM is off and mirror stowed (dark view inside SDSM scan cavity



VIIRS SD degradation



Possible culprits in VIIRS Visible Light Path



NPP VIIRS Hardware Fishbone



Contamination Fishbone



First correct hint

-----Original Message-----From: Peter D Fuqua [mailto:Peter.D.Fuqua@aero.org] Sent: Friday, December 16, 2011 12:11 AM To: Frank J DeLuccia Cc: Bruce Guenther; Waluschka, Eugene (GSFC-5510); Kennedy, Brian C. (GSFC-590.0)[ASRC MANAGEMENT SERVICES INC]; James.D.Barrie@aero.org; Chung-Tse Chu Subject: Contamination on silver mirrors

Hi Frank,

Enclosed please find charts that show that properly manufactured

- FSS99 should be robust to space exposure, that show the change in reflectivity of a protected silver mirror due to addition of a variety of hypothetical contaminants, and that addition of a tiny amount of a material with a refractive index similar to a grey metal (Cr, Ni, Ti, or Cd) can produce losses that are very similar to what is observed on orbit.
- Lastly, the SDSM mirror has a completely different design. We should not assume it will behave the same way.

Following up on your earlier questions, if we pursue this contamination hypothesis, we would expect that the contamination source is directional and that the primary mirror is only exposed to the source when the telescope is rotating. When the telescope is stowed, the source might

- deposit on the outside of the telescope housing. Furthermore, to explain
- the observations, we would also need the contamination rate when the Nadir Aperture Door opened. This might occur if the source gets warmer due to solar exposure or Earthshine.
- We're looking solarization hypotheses as well, but that didn't make it into these charts. The same logic will apply. Thin-film Interference effects may have a strong influence on the wavelengths were absorptive loss is realized.

Please distribute as you feel appropriate.

- Best Regards,
- Pete

• (See attached file: Fuqua Contam effects on protected silver v3.pptx)

Tungsten?

- Tungsten oxide found on surface of VIIRS follow-along mirror via TOF-SIMS. Estimated thickness 15-30Å
- Spectral loss on UV exposure of telescope follow along witness mirror closely matches observed VIIRS loss.
- Literature indicates tungsten oxides are photochromic in UV.
- Very few sources available to deposit tungsten or its oxide
- Very high temperature (>1000°C) needed to sublimate tungsten or an oxide.
- Volatile tungsten compounds do exist, but none are part of the spacecraft build. Deposition from these sources is also typically an elevated temperature process.
- Potential external sources:
 - Tungsten counterweight in ATMS. No line-of-sight and no mechanism.
 - Lubricants:
 - No evidence that tungsten disulfide was substituted for molybdenum disulfide in any spacecraft lubricants. Temperatures too low. No sulfides evident.
 - Vacuum roughing pumps did not use lubricants containing tungsten
 - CERES witness sample did not degrade in test, suggesting source outside of I&T.
- Potential internal sources:
 - Dry lubricant in RTA launch locks: no mechanism to convert from particles to film
 - FOD: no mechanism
 - Radiation shield on VIS-NIR detectors: too cold
- Potential workmanship sources:
 - Machining: Tungsten carbide cutting surfaces do not transfer the material to surfaces.
 - Optical coating process: Discussions with Quantum ongoing.

Tungsten source

• VisNIR cause: Quantum deviated from the qualified FSS99 coating process (on the four SNPP RTA mirrors) that resulted in the deposition of tungsten on the mirrors, with a consequence of spectral degradation on-orbit



Re: Lessons Learned for Ion Bombardment of FSS99 Silver Coating

<u>Issue Description:</u> There was a FSS99 silver coated mirror reflection degradation issue. The root cause was determined to be Tungsten Oxide film on the surface of the FSS99 coated mirrors that was introduced after the coating was applied in an attempt to increase reflection.

<u>Containment Action</u>: Upon original notification of the anomaly, records of the coating run were reviewed to determine if tungsten oxide could have been deposited during the standard FSS99 process. Additional records for other FSS99 coating runs for other programs were also located in order to determine if there were any coating deficiencies or anomalies present during these events. It was determined that there were no issues with subsequent coating runs for any specific program.

Quantum Ion Gun

- Ion gun used for substrate deposition per normal FSS99 process
- Deviation: gun used for oxygen ion bombardment to improve reflectivity

Aerospace comes to the rescue

- Chemical analysis of witness samples using Time-of-Flight Secondary Ion Mass Spectrometry (ToF-SIMS)
- Vacuum testing (UV exposure) of witness samples
- Thin film modeling of FSS99 + WOx

Aerospace presentation later in week.

Executive Summary

- Problem Upon opening the SNPP Visible Infrared Imager Radiometer Suite (VIIRS) Nadir doors on-orbit, analysis of VIIRS data revealed accelerated and continued degradation of Visible/Near Infrared (VisNIR) bands throughput. The degradation was correlated with ultraviolet (UV) exposure. A secondary degradation was observed in the same spectral region within the VIIRS Solar Diffuser Stability Monitor (SDSM).
- Cause Quantum (previously Denton) deviated from their qualified mirror coating process that resulted in tungsten oxides being deposited on the SNPP Rotating Telescope Assembly (RTA) mirrors. The SDSM degradation was traced to detector:SiPN sensitivity to high energy particles.
- Corrective action The SNPP predictions show that all bands will stay within specification. The science team will continue trending and perform calibration table adjustments to normalize the product. JPSS-1 witness samples have been tested and the results showed no spectral degradation and therefore no additional corrective action is required.
- Residual risk The residual risk is considered low as the predictions indicate the instrument will continue to perform within specifications through the life of the SNPP mission. There is no risk to JPSS-1 VIIRS.

VIIRS VisNIR/DNB & Solar Diffuser Optics

Note: reflective elements shown as transmissive elements for simplicity



Lessons learned

- Don't rush Maintaining schedule required a Friday pump down for Saturday coating run.
- Question consequences of deviations from established procedure.
- Test as you fly. Expose optics to solar spectrum in vacuum and, while still in vacuum, measure reflection in vacuum.

Backup









Key ART Personnel



- ART Lead G. Iona
- Code 500 M. Hagopian
- RTN Lead Eric Johnson
- Data Analysis F. DeLuccia V. Chiang
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VIIRS Anomaly Investigation MISSION January 17, 2012

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VNIR and SWIR Bands Still Have Healthy SNR Margins







SDSM SNR Margins Healthy, Especially in NIR



- Minimum SNR requirement is 1000:1
- Sun view provides lower signal than SD view, thus lowest SNR
- SNR calculated as: $SNR = \frac{Signal}{\sqrt{DarkNoise^2 + SignalNoise^2}}$

SNR measure noisy. Varying signal in SD view allows few samples (5) in Signal_Noise calc that is avg'd over 14 cycles



Closure Approval for Suomi National Polar-orbiting Partnership (SNPP) VIIRS



Degradation Anomaly Root Cause/ Corrective Action Investigation Report 6/14/2012

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Gregory Ayers' Date Lead Instruments Systems Engineer

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