Santa-Clara University

Extended Life Flight Results from the GeneSat-1 Biological Microsatellite Mission

Giovanni Minelli

Research Assistant - Robotic Systems Laboratory - Santa Clara University

Email: gminelli@scu.edu Web: http://rsl.engr.scu.edu/



Outline

- Introduction
- GeneSat-1
 - Overview
 - Long-Term Performance
- Ground Segment
 - Primary Station
 - Low-Cost Deployable Stations
 - MHX 2420 Upgrade
- Heritage Missions
- Conclusions



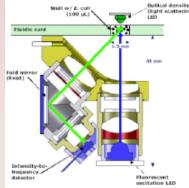
GeneSat-1 Overview

- Launch Dec. 16, 2006
 - Secondary payload on Minotaur launch from WFF
 - Developed by NASA Ames Research Center
- Unique biological experiment
 - The study of microgravity effects on E. coli metabolism
 - GFP and well turbidity were primary measurements
- Technology validation demonstration
 - Small satellites can be a valuable asset in conducting tests in orbit
 - Miniaturization of biological support system including fluidics, heating and optics
- Mission Success 1 Month of Operations
 - SCU operations team responsible for ground segment and extended mission after NASA handover
 - At least 5 heritage missions flying GeneSat-1 bus and/or payload elements, motivating long-term performance trend analyses



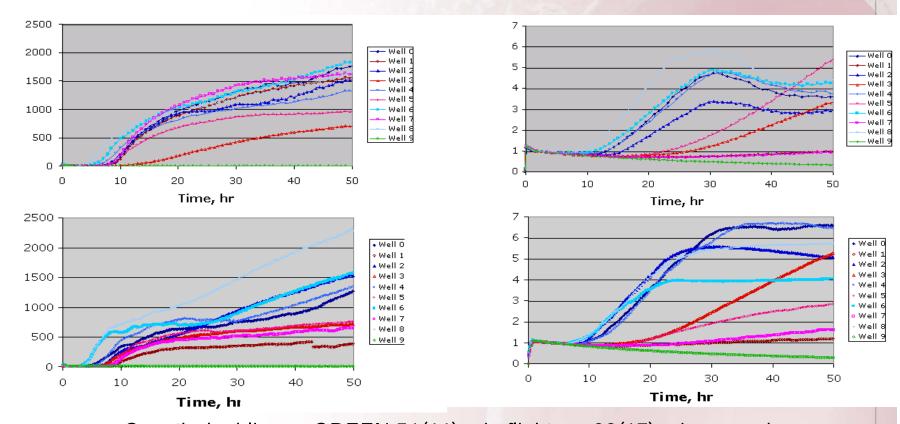








GeneSat-1 Biology Results



Growth doubling: pGREEN 51(11) min flight vs. 38(17) min ground AcGFP 45(6) min flight vs 33(5) min ground



Long Term Performance – Bus

Nominal Overall Performance

Telemetry measured at regular intervals over 15 months

Power

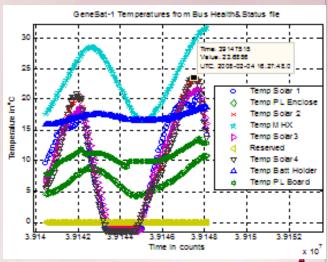
- Solar panel degradation 2.62%/ year
- Voltages nominal

Thermal

- Thermal profile remains constant
- Microhard radio warmest component at 35.8 °C

CPU

- No indicated resets, latch-ups, or SEUs
- Experiment data successfully downloaded after 15 months of storage
- Clock drift >24 hrs over 15 months
 Minelli SmallSat Conference 2008



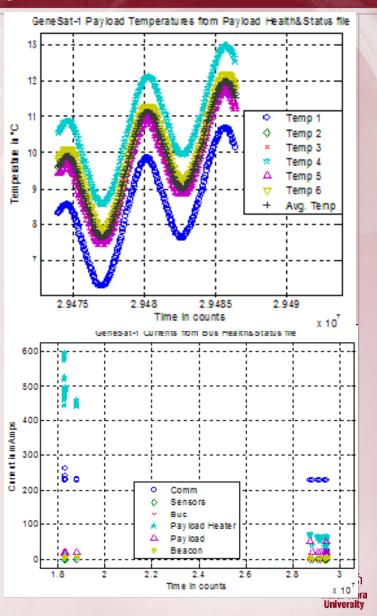


Long Term Performance - Payload

Mission Success

Heater Anomaly After ~10 Months

- Discovered when setpoint was not maintained. Well plate temp fluctuated with orbital period.
- Current draw dropped from 500-700 mA to 50 mA.
- Subsequent heater shut off
- Suspected blown FET on duty-cycled well plate heater



Primary Station

• 18m parabolic antenna

- Owned by SRI International, land leased from Stanford University
- Recently refurbished to support 2.4 GHz

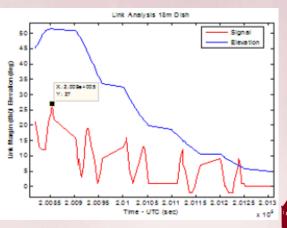
Satellite command and telemetry link

- Microhard MHX-2400: low cost ISM band 2.4 GHz transceiver
- COTS radio not originally intended for space flight
- GeneSat-1 first flight demonstration for unit

Long term performance

- Link strength measured over time no noticeable signal degradation.
- Satellite rotation created signal 'coning', which occasionally affected link performance





Deployable Stations

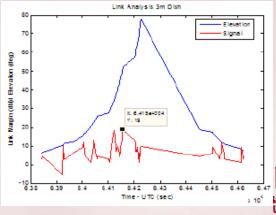
Two stations developed

- Support of NASA PreSat and NanoSail-D missions
- Installed in El Salvador and Kwajalein Atoll
- Link analyses showed feasibility of such stations for limited operations (~10 dB loss vs. SRI dish).
- COTS equipment including 3m parabolic antenna
- 0.5° pointing accuracy, 6°/sec azimuth rate
- Use of OSCAR dual Yagi antennas for amateur radio beacon tracking

Validation

- Used for GeneSat-1 contacts, 15-70° elevation
- Weather, temperature, noise, and coning effects more noticeable







MHX 2420

Upgrade from 2400 to 2420

- Microhard Inc. discontinued MHX-2400 unit used on GeneSat-1 flight unit
- MHX-2400 spares increasingly difficult to find
- Enhanced MHX 2420 offered with backwards compatibility after factory installed firmware modification

Validation

- Bench tests confirmed backwards compatibility
- MHX-2420 successfully used in GeneSat-1 contact when installed at SRI primary ground station (7/24/08 - 30+ pages downloaded)







Heritage Missions

GeneSat-1 Heritage

 Mission success spawned a number of follow-on missions adapting key components of GeneSat-1 bus and payload

PharmaSat (NASA ARC)

- PI driven pharmacological experiment
- Bus reflight. Larger, more complex biological support payload.
- Secondary payload on WFF launch (Oct. 2008)

PreSat (NASA ARC)

 Technology validation and evaluation flight for PharmaSat (SpaceX Falcon-1 Launch, Aug. 08)

NanoSail-D (NASA MSFC/ARC)

 GeneSat-1 bus with one-of-a-kind deployable solar sail payload. (SpaceX Falcon-1 Launch, Aug. 08)

Additional

- PI-driven biological laboratory satellites (OOREO Dec. 09)
- NASA Ames' COTSAT program (Mar. 09)
- Future research and industry partners



PreSat/ PharmaSat



NanoSail-D



Conclusions

GeneSat-1 Heritage

- Mission proved that triple CubeSat configurations are effective for in-situ biological research
- Designs intended for a short-lived study can last successfully for extended periods of time
- New flights carrying a number of heritage components
- Streamlined ground segment development for heritage mission support

Educational Contributions

- Small satellites in the classroom: student operations laboratory, space systems design class, university nanosatellite program, publications, undergraduate and master's level theses
- Worldwide amateur HAM radio community involved in beacon packet collection and submission
- Student involvement in other institutions, i.e. University of Central America, FL Salvador



Acknowledgments

NASA Ames Research Center

 John Hines, Elwood Agasid, Charlie Friedricks, Matthew Piccini, Macarena Parra, Linda Timucin, Mike Henschke, Antonio Ricco, Ed Luzzi, Nghia Mai, Mike McIntyre, Robert Ricks, David Squires, Chris Storment, John Tucker, Bruce Yost, Greg Defouw

• Santa Clara University

 Christopher Kitts, Karolyn Ronzano, Christopher Beasley, Mike Rasay, Ignacio Mas, Phelps Williams, Paul Mahacek, Brian, Gamp, John Shepard, Jose Acain

SRI International

 Michael Cousins for primary ground station support

University of Central America

 Ismael Sanchez and Oscar Valencia for their help in El Salvador ground station setup

Kwajalein Ground Station

Steve Buckley, SpaceX, RTS, KRS



Santa-Clara University

QUESTIONS?

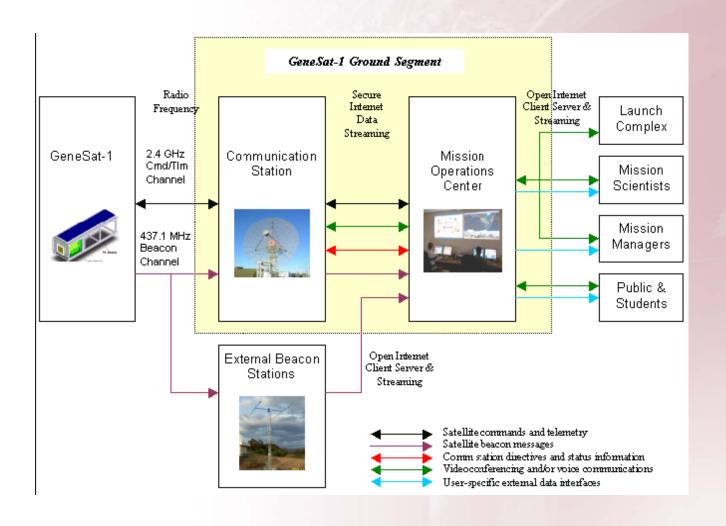


Link Budget

GeneSat-1 ISM Cmd&Tlm			
2.4GHz DownLink Budget			
Item	Sym	Units	DL
Orbit Altitude (km)		km	450
Elevation Angle		deg	10
Frequency	f	GHz	2.4
Transmitter Power	Р	Watts	1
Transmitter Power	Р	dBW	0
Transmitter Line Loss	LI	dBW	-1
Avg Transmit Antenna			
Gain	Gpt	dBi	3.0
Transmit Total Gain	Gt	dB	2.0
Eq. Isotropic Radiated			
Power Propagation Path	EIRP	dBW	2.00
Length	s	km	1570
Space Loss	Ls	dB	-164.0
Propagation and	Lo	ub.	-104.0
Polarization Loss	La	dB	-3
Receive Antenna Diameter	D	М	3
Receive Antenna Eff	Eta		0.55
Peak Receive Antenna			0.00
Gain	Grp	dBi	34.96
Receive Antenna Line			
Loss Receive Antenna	Lr	dB	-0.5
Beamwidth*	Theta	deg	2.92
Receive Antenna	111010	aog	2.02
Pointing Error	Е	deg	0.50
RX Antenna Pointing			
Error Loss	L_θ	dB	-0.35
Receive Antenna Gain with pointing error	Gr	dB	34.1
System Noise	- 01	UD.	34.1
Temperature **	Ts	K	585
Data Rate	R	bps	86000
Eb/No (1)	Eb/No	dB	20.7
Bit Error Rate	BER		10-5
Required Eb/No (2)	Req Eb/No	dB-Hz	13.5
Implementation Loss (3)		dB	-2
Margin		dB	5.2
margin		UD	J.Z



Ground Segment





Ground Segment Architecture

