

# Performance Assessment of Single and Dual-Frequency, Commercial-based GPS Receiver for LEO orbit

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# Background

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- Previous GPS receivers
  - Many of GPS receivers operate on L1 frequency and its navigation accuracy is limited by ionospheric path delay.
  - Earlier studies proposed the correction method for its delay using an ionospheric model for LEO altitude.
- Dual Frequency GPS receivers
  - Direct correction of ionospheric path delay.
  - The onboard navigation accuracy is much better than that of single frequency GPS receivers.
  - Limitaion on available selection of space-capable one.

We've adopted the low cost commercial dual frequency GPS receiver  
NovAtel "OEM4-G2L"



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# Description of NovAtel OEM4-G2L

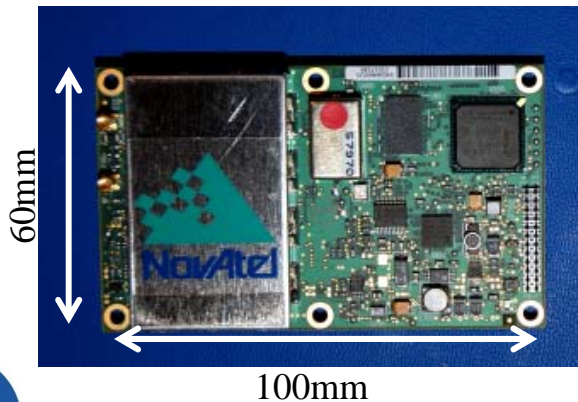
## ➤ Remarkable points

- Small size, weight, and low power consumption
- 24 tracking channel (12ch for L1 C/A & 12ch for L2 P-code frequency)

## ➤ Firmware modification

- Removal of altitude and velocity limitation and correction of tropospheric delay.

(The study\* of “OEM4-G2” reports the large position error of 13m with tropospheric delay correction. \*O.Montenbruck, DLR)



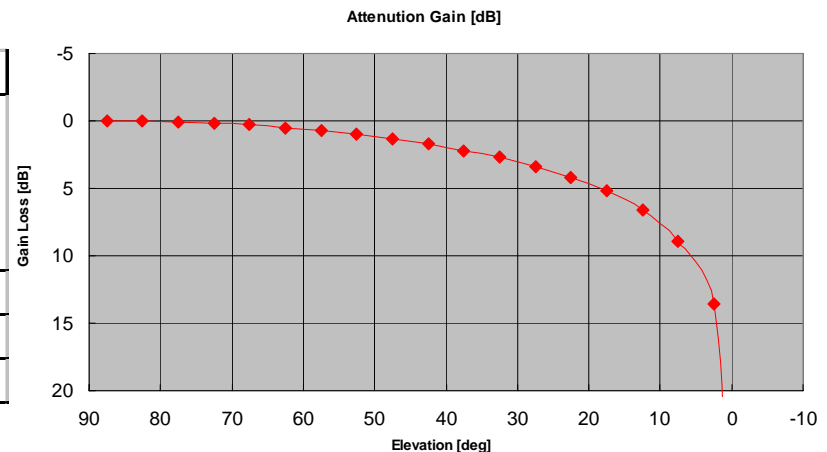
item	specification
size[mm]	60 x 100 x 16
Weight [g]	56
Power [W]	1.6W @ 3.3V
Frequency	1575.42MHz (L1) 1227.60MHz(L2)
interface	RS232, RS422, TTL, PPS
Data rate [Hz]	20

# Performance Test of OEM4-G2L

- GPS signal simulator : Spirent STR4760
- Test item :
  1. Initial acquisition test
  2. Error free scenario test
  3. Ionospheric error scenarios (constant TEC model)
    - case1 : TEC value =  $1e17$  electron/m<sup>2</sup>
    - case2 : TEC value =  $1e18$  electron/m<sup>2</sup>

Common setting of simulator

ITEM	Setting
Orbit	Sun Synchronous Orbit Altitude : 650km Inclination : 97.99deg semi-major axis : 7028km
tropospher model	OFF
GPS constellation position error	Disable
GPS clock divergence	Disable



# Initial Acquisition Performance Test

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## ➤ Objective

To evaluate the Time To First Fix (TTFF) from cold start.

## ➤ Simulation configuration

- 12ch output
- Ionosphere model : constant TEC ( $1.0e17$  electron/m<sup>2</sup>)

## ➤ Results

- TTFF is about 2 to 8min. ( 5 to 40min @ MGPSR)
- OEM4-G2L is well able to operate under low earth orbit

Case	TTFF [sec]	Latitude at receiver activation [deg]
1	315	-0.03314
2	118	75.76264
3	497	26.62787
4	115	-47.9419
5	111	-55.8707



# Performance Assessment (Error Free Scenario Test)

## ➤ Objective

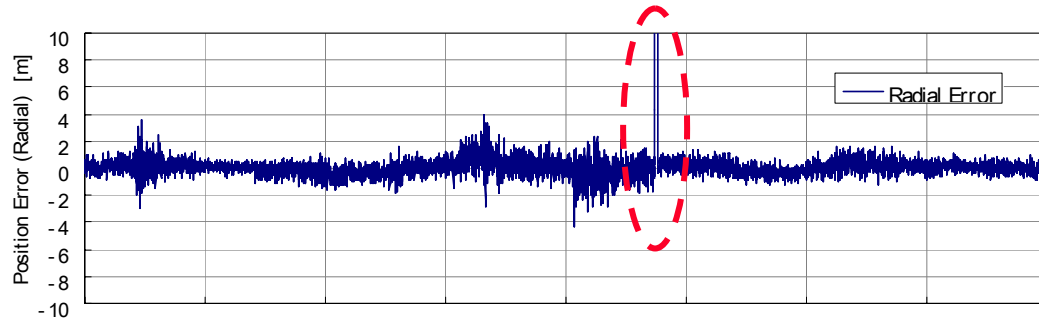
To provide the reference data for the other test cases and verify the effect of the removal of tropospheric delay correction on the firmware.

## ➤ Simulation configuration

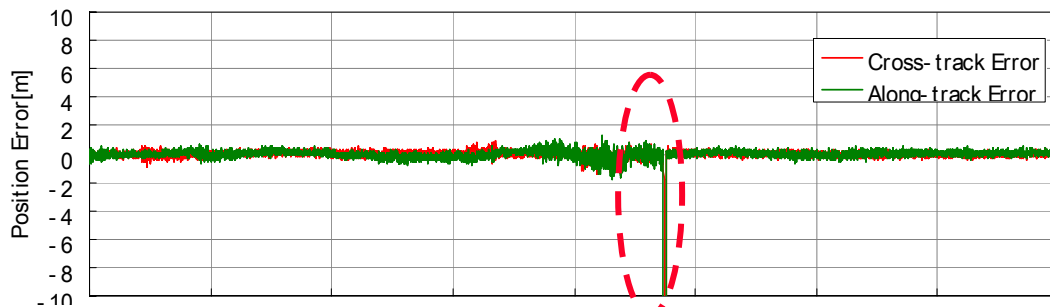
- 8ch output
  - Ionospheric error : OFF
- Error sources are receiver clock and measurement noise.



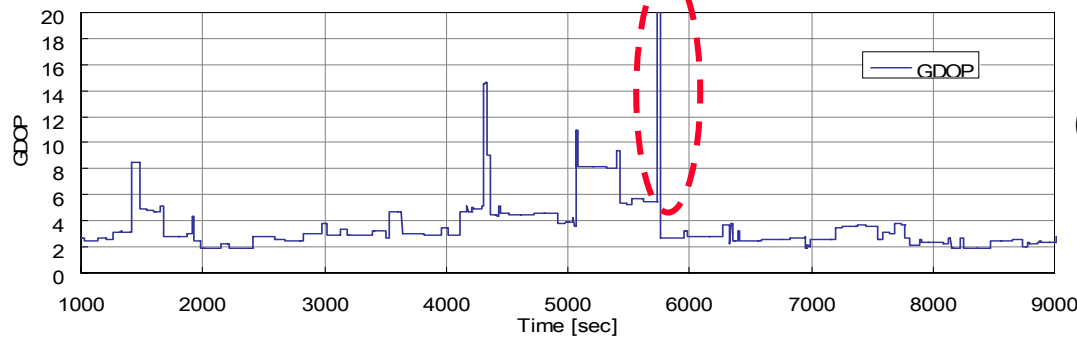
# Error Free Scenario (Position Error)



(a) Radial error



(b) Cross and Along Track error



(c) GDOP



# Summary of Error Free Scenario

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- Removal of Tropospheric delay lead to good accuracy
- Navigation accuracy is deteriorated when GDOP is high.  
→ Antenna layout and filter design should be considered carefully.

Summary of the results

	Radial	Cross-track	Along-track
Mean Position Error	0.0258 [m]	-0.0127 [m]	-0.00919 [m]
Position Error S.D.	0.571 [m]	0.186 [m]	0.252[m]
Mean velocity Error	0.0579 [m/s]	0.00264[m/s]	0.00498[m/s]
Velocity Error S.D.	0.0766[m/s]	0.0282[m/s]	0.0322[m/s]

※S.D. : Standard Deviation



# Ionospheric Error Scenario Test

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## ➤ Objective

To evaluate the effectivity of the dual frequency (L1 & L2) observation to remove the ionospheric delay.

## ➤ Simulation configuration

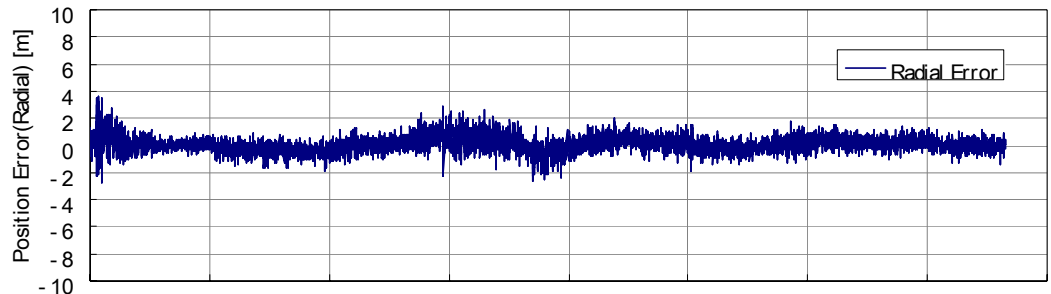
- 8ch output
- Ionospheric model : constant TEC ( $1.0e17$  electron/m<sup>2</sup>)



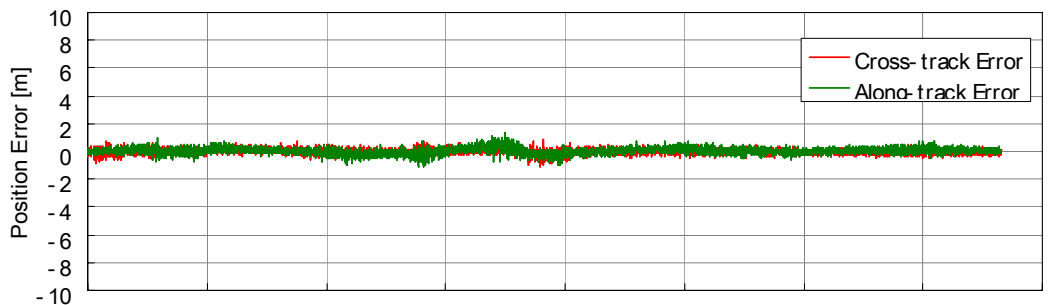
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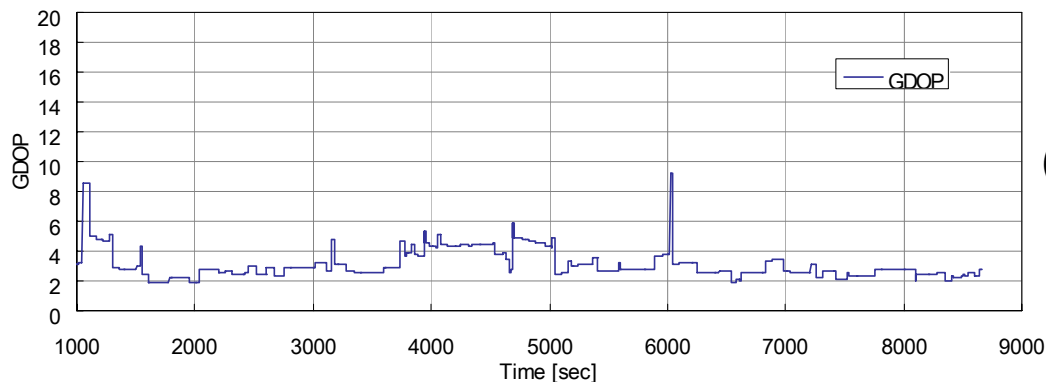
# Ionospheric Error Scenario (position error)



(a) Radial error



(b) Cross and Along Track error



(c) GDOP

# Summary of Ionospheric Error Scenario

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- Navigation accuracy is still good even in ionospheric path delay.
- Dual frequency observation is well able to remove the ionospheric path delay.

## Summary of results

	Radial	Cross-track	Along-track
Mean Position Error	0.0624 [m]	-0.00982[m]	-0.0292 [m]
Position Error S.D.	0.584 [m]	0.186 [m]	0.247[m]
Mean velocity Error	0.0604 [m/s]	0.00039[m/s]	-0.0091[m/s]
Velocity Error S.D.	0.0744[m/s]	0.0229[m/s]	0.0256[m/s]

# High TEC Scenario Test

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## ➤ Objective

To evaluate the effect of the large ionospheric delay on the GPS receiver.

## ➤ Simulation Configuration

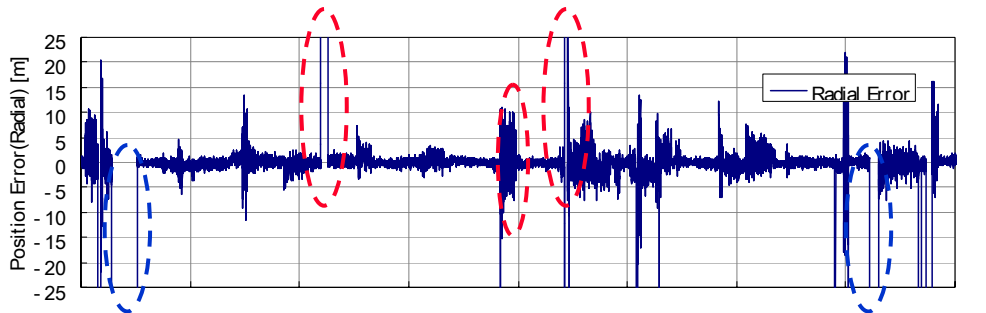
- 8ch output
- ionosphere model : constant TEC ( $1.0e18$  electron/m<sup>2</sup>)



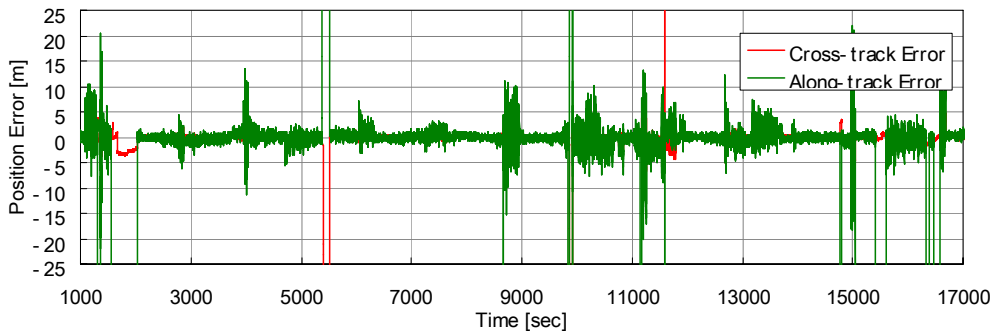
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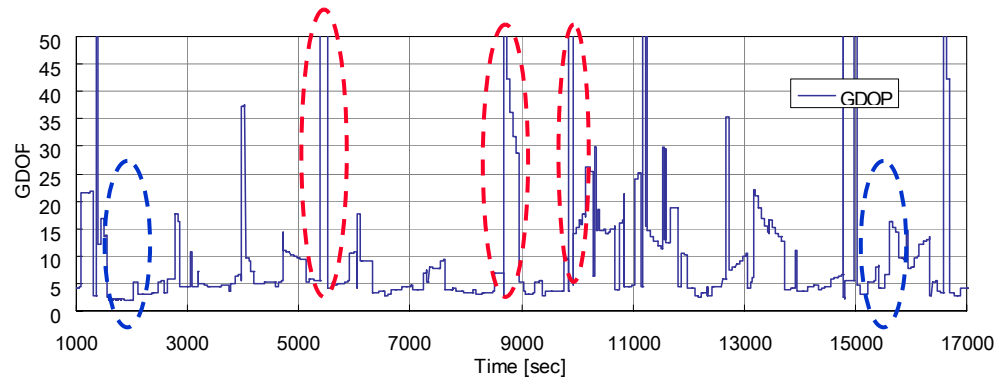
# High TEC Scenario (Position error 1/2)



(a) Radial error

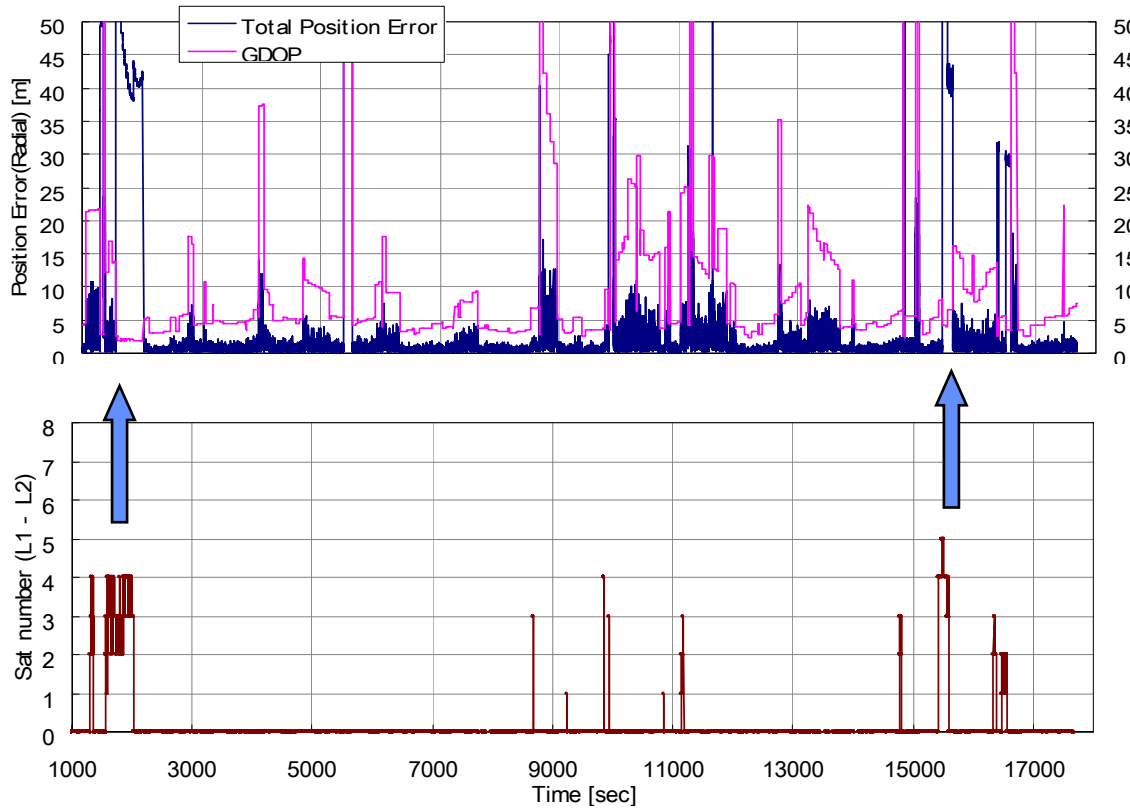


(b) Cross and Along Track error



(c) GDOP

# High TEC Scenario (Position Error 2/2)



Position error remarkably **increased with loss of L2 signal**

# Summary of High TEC scenario

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- Navigation errors are mainly due to poor DOP.
- But..., some data shows large position error in relatively good DOP.
- These position errors are due to loss of L2 signal
  - OEM4-G2L switches to the model based ionospheric delay correction.
  - The model (Klobcar model) used in this case does not suit to space use.

Summary of results

	Radial	Cross-track	Along-track
Mean Position Error	0.00479 [m]	-0.00753[m]	-0.141 [m]
Position Error S.D.	1.06[m]	0.294 [m]	0.549[m]
Mean velocity Error	0.124 [m/s]	0.0129[m/s]	-0.080[m/s]
Velocity Error S.D.	0.179[m/s]	0.0581[m/s]	0.109[m/s]



# JAXA Micro-GPS Receiver (MGPSR)

- The COTS based single frequency GPS receiver
- Firmware modification
  - Carrier phase output is added.
  - removal of altitude limit, Tropospheric and Ionospheric delay correction

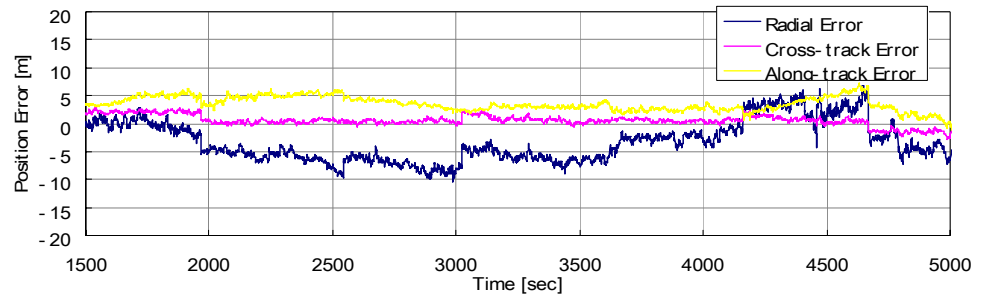


Item	Specifications
Size	72 x 50 x 40 mm
Mass	215g
Power	1.5W (typical)
Frequency	1575.42 MHz (L1)
No. of channels	8 ch
Output data	PPS signal Navigation data Raw data Ephemeris data
Interface	RS-422, +5VDC

# Ionosphere Scenario Tests (compared with OEM4-G2L)

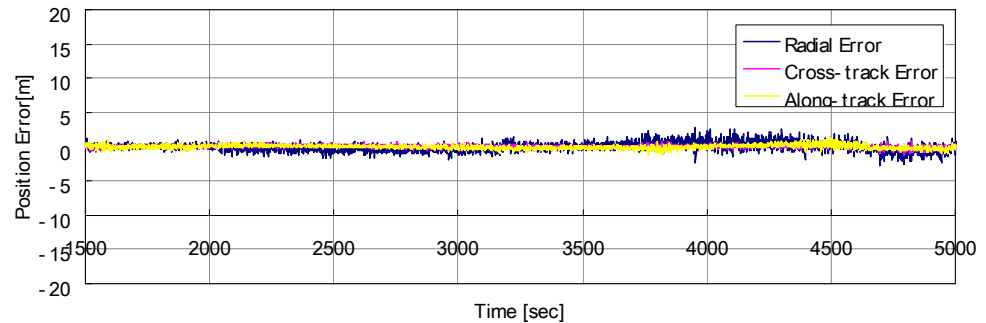
- Navigation accuracy of MGPSR is affected by the ionospheric delay especially in radial direction.

	Radial	Cross-track	Along-track
Mean Position Error	-2.95[m]	0.542[m]	3.41[m]
Position Error S.D.	3.75[m]	0.956[m]	1.39[m]



Results of MGPSR

	Radial	Cross-track	Along-track
Mean Position Error	0.0624 [m]	-0.00982[m]	-0.0292 [m]
Position Error S.D.	0.584 [m]	0.186 [m]	0.247[m]



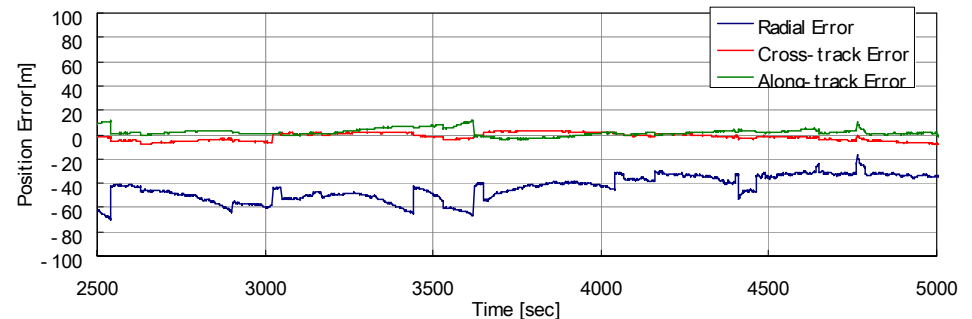
Results of OEM4-G2L



# High TEC Scenario Test (compared with OEM4-G2L)

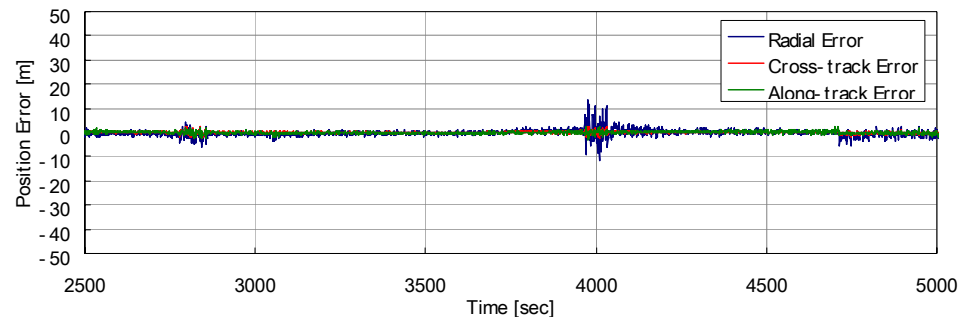
- Ionospheric delay affect the navigation error of the MGPSR more significantly than that of OEM4-G2L.

	Radial	Cross-track	Along-track
Mean Position Error	-46.8 [m]	-2.98[m]	1.84[m]
Position Error S.D.	11.3 [m]	3.57[m]	2.89[m]



Results of MGPSR

	Radial	Cross-track	Along-track
Mean Position Error	0.00479 [m]	-0.00753[m]	-0.141 [m]
Position Error S.D.	1.06[m]	0.294 [m]	0.549[m]



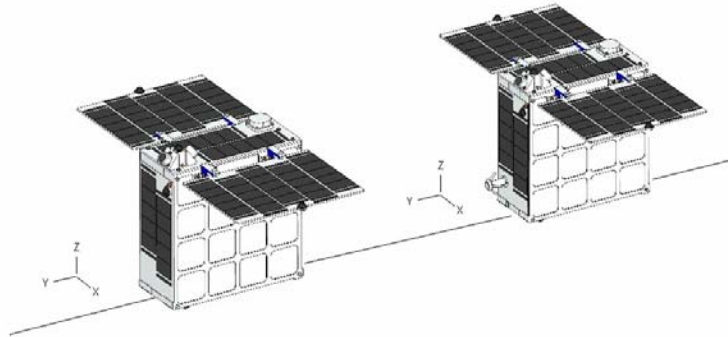
Results of OEM4-G2L



# JC2Sat-FF Mission

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- JC2Sat-FF Mission
  - International joint research between CSA and JAXA.
  - A nanosat formation flying mission based on differential drag technique and GPS based relative navigation.
  - The orbit information of each nanosat is obtained by its respective GPS receiver. This information is then transmitted via a S-band antenna to the other nanosat.
  - On-orbit Attitude & Orbit Determination Systems software will then compute the separation between the nanosats and then determine the needed control authority to maintain the baseline.



# SUMMARY

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- ◆ Simulation based performance assessment of a commercial dual frequency GPS receiver NovAtel “OEM4-G2L” is performed. The result shows ;
  - OEM4-G2L shows well initial acquisition performance in LEO.
  - Under the condition in which Ionospheric delay is correctly removed by the dual frequency observation, the OEM4-G2L provides accurate navigation solution.
- ◆ The comparison of the performance between OEM4-G2L and JAXA MGPSR shows some advantages of the dual frequency receiver.
- ◆ More detailed study such as assessment of raw measurement accuracy will be performed in the future works.

