# ELECTROCHROMIC RADIATORS FOR MICROSPACECRAFT THERMAL CONTROL

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## Introduction



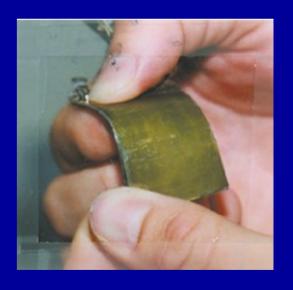
- Thermal control challenges for small spacecraft
  - insufficient thermal mass
  - limited electrical power for survival heating
- Thermal insulation (e.g. MLI blanketing) is the most straightforward approach, but not always the solution



- Mechanical Louvers
  - used to modulate heat rejection from radiators
  - bulky mechanical devices, difficult to miniaturize
  - often opened and closed via a bi-metallic thermostatic actuator with a single temperature set point
- Thin-film variable-emittance coatings offer the functionality of mechanical louvers but with decreased mass, cost, and mechanical complexity



**Mechanical Louvers** 



Variable-emittance Coating



## Variable-Emittance Radiators



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- A controlled variation of the Infra-red emittance of a radiator can maintain the temperature or heat rejection in response to changing environment
- Effect of variability of infrared emissivity of a flat plate radiator with no solar irradiation:

Stefan-Boltzmann Law: Q = AεσT<sub>1</sub><sup>4</sup>

- Constant Heat Rejection:  $T_2/T_1 = (\epsilon_1/\epsilon_2)^{0.25}$ 

- Constant Temperature:  $Q_2/Q_1 = \epsilon_2/\epsilon_1$ 

- Several variable-emittance technologies are being developed by DOD and NASA sponsored Small Business Innovation Research (SBIR) grants for potential space applications:
  - Microelectromechanical (MEMS) machined microlouvers
  - Electrophoretic and Electrostatic surfaces
  - Electrochromic Coatings

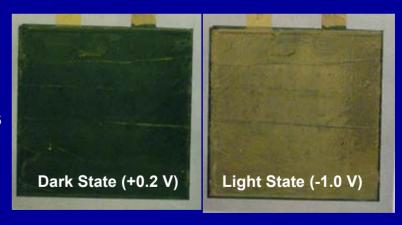


# **Thin-film Electrochromic Coatings**



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- Electrochromics are electroactive materials that show a reversible optical property change when an electric field is applied.
- Visible-NIR region: (0.4 to 1.1 μm); IR Region: (2 to 45 μm)
- Reflective and/or transmissive characteristics of the film are changed by the application of a small activation electrical potential (usually DC, <± 5 V.)</li>
- Infra-red modulation:
  - Dark state = highly IR-absorbing
  - Light state = IR-transparent
- Potential for active and continuous modulation of radiator emissivity values
- Requires minimal electrical power to operate (depends upon leakage current)



**Electrochromic Devices (Ashwin-Ushas, Inc.)** 



# Conducting Polymer-based Electrochromics Polymer-based



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- Ashwin-Ushas Corp. / NASA-JPL
- Devices are composed of a number of layers similar to the anode, cathode, and electrolyte in a battery
  - Active CP layer undergoes electrochemically-induced oxidation and reduction with an applied voltage.
  - Completely reduced state is IRtransparent. Partially oxidized state is highly IR-absorbing.
  - Ionic Electrolyte liquid from -100°C to + 280°C
- Performance:
  - Emissivity change: ~0.55
  - Emissivity limits tailorable: 0.15 to 0.85
  - Solar Absorptance < 0.29
- Thin (< 0.5 mm), flexible panel construction
- Light weight: ~ 0.8 kg/m^2
- Low power consumption:
  - Peak Transient ~4 mW/cm2 for < 30 sec
  - Steady-state: < 40 µW/cm2

**Conducting Polymer** Au Layer Microporous membrane **Conducting Polymer** Au Layer Membrane

Front (Working) **Electrode** 

**Back (Counter) Electrode** 







# **Microspacecraft Application**



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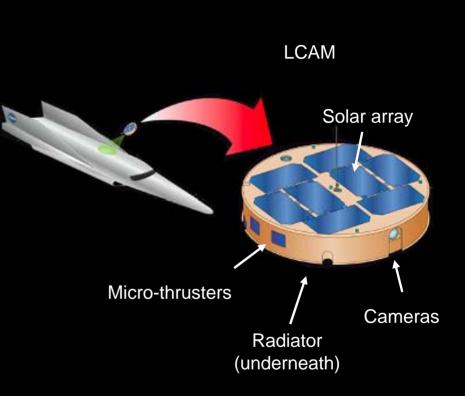
- Micro-Inspector is a deployable, mobile camera platform intended for inspection of exterior surfaces of a host spacecraft
- Based on the Low Cost Adjunct
   Microspacecraft (LCAM) architecture

#### Attributes:

- less than 3 kg and 25 cm<sup>3</sup>
- solar array and battery powered
- low pressure, liquid butane-based propulsion system

#### Thermal Control Requirements:

- maintain all avionics/instruments/batteries within allowable temperatures
- utilize waste heat from avionics and instruments to vaporize propellant
- manage the waste heat so that the butane propellant does not become overpressurized





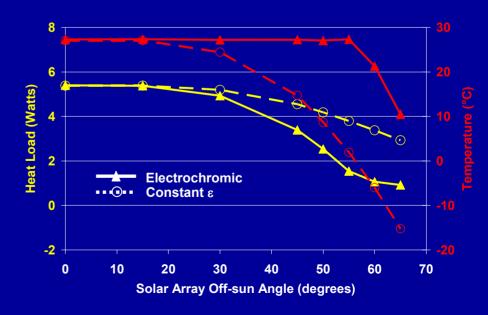
# Micro-Inspector Electrochromic Radiator



- Electrochromic Device (ED) from Ashwin-Ushas to be applied to the underside thermal radiator
- Device consumes less than 20 mW steady-state power
- High emissivity state ( $\varepsilon$ =0.7) allows for rejection of waste heat loads
- Low emissivity state ( $\varepsilon$ =0.15 to 0.3) conserves waste heat for butane vaporization
- Thermal modeling indicates Electrochromic radiator extends steady-state operation modes for large off-sun angles



**Micro-Inspector Thermal Modeling Results** 





# Electrochromics Development Testing



- **Ashwin-Ushas Electrochromic devices are** currently being developed for increased reliability
- NASA SBIR program: environmental exposure (gamma radiation, UV, solar wind, atomic oxygen, hard vacuum, thermal cycling)
- Micro-Inspector development entails testing a number of devices for performance, material compatibility, manufacturability, and durability.
  - Calorimetric and reflectometer performance testing
  - Iso-butane propellant chemical compatibility tests
  - Long term storage in vacuum, openatmosphere, and gaseous Nitrogen environments
  - System thermal vacuum performance testing





# **Summary/Future Work**



- Conducting Polymer Electrochromic Devices from Ashwin-Ushas Corporation are being developed for microspacecraft thermal control applications
- An Electrochromic radiator is currently baselined as the primary thermal control device for the Micro-Inspector spacecraft project at JPL
- A technology development program is underway to assess and improve the performance, manufacturability, material compatibility, and lifetime of the electrochromic device technology.
- Further development testing as part of the Micro-Inspector project will include vibrational tests and thermal cycling tests to gauge environmental stress and system thermal vacuum tests to validate performance.
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