# Calibration Acquisitions of the Moon by CLARREO Pathfinder

Thomas C. Stone<sup>1</sup>, Constantine Lukashin<sup>2</sup> and Greg Kopp<sup>3</sup>

<sup>1</sup>U.S. Geological Survey, Flagstaff, AZ <sup>2</sup>NASA Langley Research Center, Hampton, VA <sup>3</sup>LASP, University of Colorado, Boulder, CO

CALCON Technical Conference Logan, Utah 23 August 2017



#### **Introduction: CLARREO mission**

- CLARREO = Climate Absolute Radiance and Refractivity Observatory
- Calibration Reference Spectrometers for national and international climate observing systems in orbit (Tier 1 Decadal Survey Mission)
- Reflected solar and infrared spectrometers enabling calibration across the entire solar and infrared spectrum of space instruments (e.g. CERES, VIIRS, CrIS, Landsat, geostationary imagers and sounders)
- Advance accuracy a factor of 5 to 10 for observing global climate change from space (e.g. climate sensitivity/cloud feedback)
- Higher accuracy enables earlier more accurate predictions of future climate change (15 to 20 years)
- Reduced prediction uncertainty leads to better societal decisions and economic outcomes.
- CLARREO will be a critical calibration anchor of the first international climate observing system (currently we have none)
- Economic value of a more accurate climate observing system using CLARREO as the example estimated at \$10 Trillion U.S. dollars (Cooke et al., 2014). ROI \$50:\$1 if triple global climate observation effort (true climate observing system)





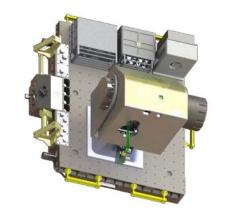




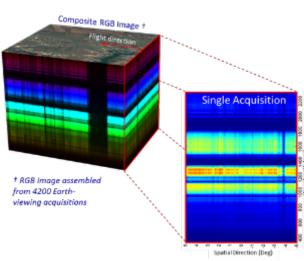
#### **Introduction: CLARREO Pathfinder Mission**

- CLARREO Pathfinder will demonstrate essential measurement technologies for the Reflected Solar portions of the CLARREO Tier 1 Decadal Survey Mission
  - Demonstrate on orbit, high accuracy, SI-Traceable calibration
  - Demonstrate ability to transfer this calibration to other on-orbit assets
- Formulation, implementation, launch, operation, and analysis of measurements from a Reflected Solar (RS) Spectrometer, launched to the International Space Station (ISS)
- Category 3 / Class D Mission, nominal 1-year mission life
   + 1 year science data analysis
- Targeted for launch to ISS in early CY2021

CLARREO Pathfinder is not the end, it is a critical step along the way to a full CLARREO Mission.



**Instrument Payload** 



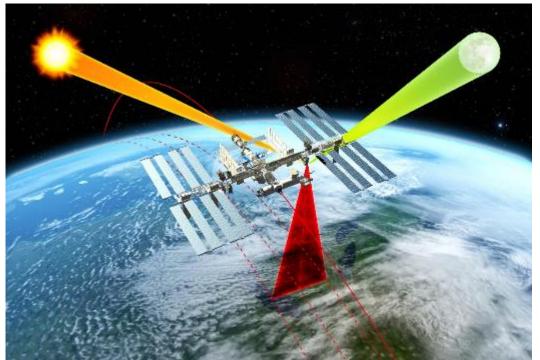
**Spectrally-Resolved Earth Reflectance** 





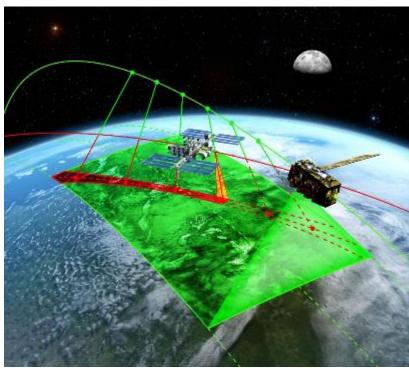
## **CLARREO Pathfinder** — Baseline Mission Objectives

#### Demonstrate high accuracy SI-Traceable Calibration



Objective #1: Demonstrate the ability to conduct, on orbit, SI-Traceable calibration of measured scene spectral reflectance with an advanced accuracy over current on-orbit sensors using a reflected solar spectrometer flying on the International Space Station.

#### **Demonstrate Inter-Calibration Capabilities**



Objective #2: Demonstrate the ability to use that improved accuracy to serve as an in orbit reference spectrometer for advanced intercalibration of other key satellite sensors across the reflected solar spectrum (350-2300 nm).







## **CLARREO Pathfinder (CPF) Specifications**

- Spectrally resolved Earth reflectance (350 2300 nm) measurements
  - SI-Traceable, referenced to spectral solar irradiance
  - Uncertainty requirement:  $\leq 0.3\%$  (k=1) baseline,  $\leq 0.6\%$  (k=1) threshold Baseline requirement is within a factor of 2 of full CLARREO Tier-1 Decadal Survey Mission Requirements
- On-orbit inter-calibration with CERES/RBI short wave channel and VIIRS reflectance bands
  - Uncertainty requirement:  $\leq 0.3\%$  (k=1) baseline,  $\leq 0.6\%$  (k=1) threshold Threshold requirement is a factor of 2 (CERES) to 4 (VIIRS) better than current capabilities

Prototype Instrument: LASP HyperSpectral Imager for Climate Science

Spatial Resolution: 2.5 arcmin (HySICS)

Field of View (cross-track): 10°

IFOV: 0.2°

Wavelength Range: 350–2300 nm

Wavelength Resolution: 6 nm, constant, Nyquist sampled

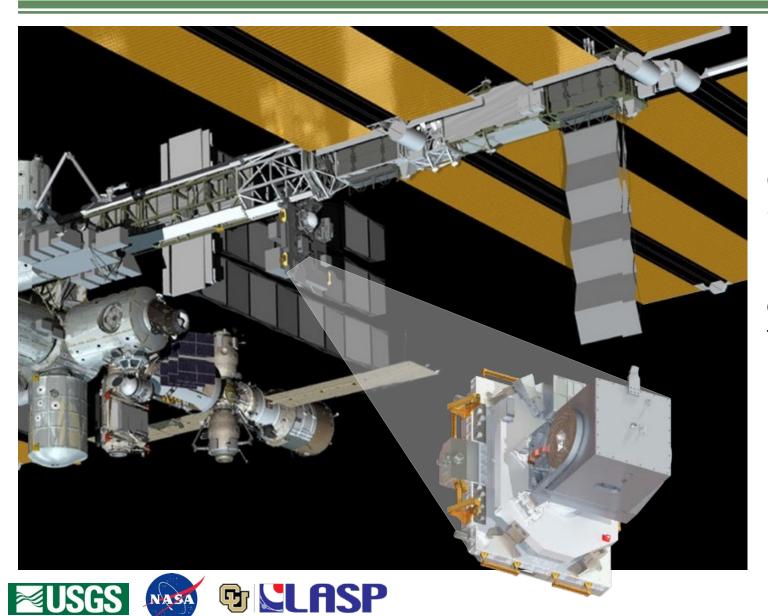
Nominal frame rate: 15 Hz







## **CPF Reflected Solar (RS) Instrument Accommodation on ISS**



ExPRESS
Logistics
Carrier ELC-1
Site 3:

ISS port side nadir location, outboard-facing pallet

# RS Instrument Field of Regard: Pointing for Solar, Lunar, Inter-calibration

Accommodations on ISS at Express Logistics Carrier #1 (ELC-1) Site 3 provides adequate viewing to meet CLARREO Pathfinder mission objectives.

Pitch

Roll Wake -105° **Port** Ram 100° Starboard +x ← -50° Accommodation studies by the NASA LaRC 0° Engineering team: J. Leckey, C. Boyer, T. Jackson **+**y ← Gimbal configuration: pitch - roll Approximate gimbal range of motion at ISS ELC-1 Site 3.







Not all pointing angles are available due to ISS accommodation.

#### **CPF RS Views of the Moon**

On ISS, the RS spectrometer slit will be oriented perpendicular to the roll or elevation gimbal axis

- aligned with the ISS ±Y body axis (starboard/port) when nadir-viewing
- to accommodate primary mission objective: pushbroom imaging of Earth

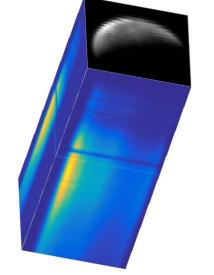
Two scanning modes will be used for RS lunar views:

- along-slit scans
  - to sample the same slice of the Moon with all detectors in the spatial direction, for flatfielding
  - accomplished with elevation (roll) gimbal movements
- across-slit scans
  - to build spectral images, for lunar radiometry
  - accomplished with azimuth (pitch) gimbal movements
  - CPF applications of lunar radiometry:
    - inter-calibration with other instruments, using the Moon as a common reference target
    - contributing to a database of high-accuracy lunar radiometric measurements, to refine and constrain the lunar calibration reference









## **Challenges for CPF RS Lunar Radiometry**

## **Spatial Sampling**

Typical lunar calibration activities utilize the spatially integrated quantity of spectral irradiance. For imaging instruments, this involves summing radiance pixels over the Moon image:

$$E_{\lambda} = rac{1}{\eta}\,\sum\limits_{i=1}^{N_{
m p}} \Omega_i\,L_{i,\lambda}$$

 $\eta = ext{oversampling factor}$   $\Omega_i = ext{pixel solid angle}$   $L_{i,\lambda} = ext{pixel radiance}$   $N_{ ext{p}} = \# ext{ of pixels on Moon}$ 

For reliable lunar irradiance measurements, a critical parameter is accurate knowledge of the oversampling of the Moon disk

The 2-axis gimbal of CLARREO Pathfinder, combined with the fixed orientation of the slit, means across-slit scanning (by azimuth slewing) will trace a curved path for off-nadir view directions.

Thus for off-nadir across-slit scans, the slit projection on the target will rotate with the azimuth gimbal movement.







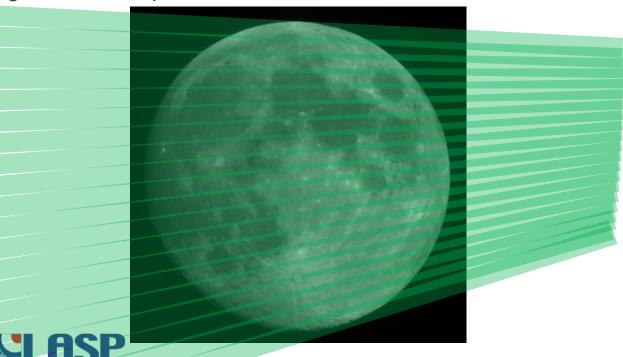
## **Spatial Sampling Issue for CPF RS Lunar Radiometry**

The rotation of the slit projection on the target means spatial sampling will be non-constant, and dependent on the spatial pixel position.

To determine oversampling of the Moon disk will require:

- knowledge of the slew rate, sampling rate, and the elevation gimbal angle
  - > to determine the geometric oversampling for each spatial pixel
- knowledge of the location and extent of the Moon slice in each scan frame
  - > to map the target onto the spatial dimension

Simulated curved scan showing differential oversampling of the Moon disk





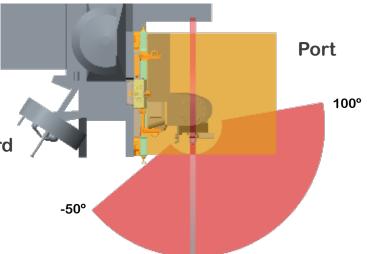


## **Challenges for CPF Lunar Radiometry**

### Moon Viewing Constraints

The instrument location on the nadir side of the ISS restricts views above the horizon.

- starboard-side views will not extend to space
- port-side views will be partially obstructed by the ISS truss above
   100° from nadir Starboard
- moveable objects (e.g. solar panels)
   will cause temporary obstructions



A simulation of potential Moon view opportunities provided preliminary data on possible lunar acquisitions

- ISS in stable circular orbit at 51.6° inclination, LVLH attitude: +XVV
- space-viewing constraints imposed as limits on gimbal motions
- no accommodation for temporary obstructions or attitude variations
- simulation was run for one-year flight (nominal CPF mission)







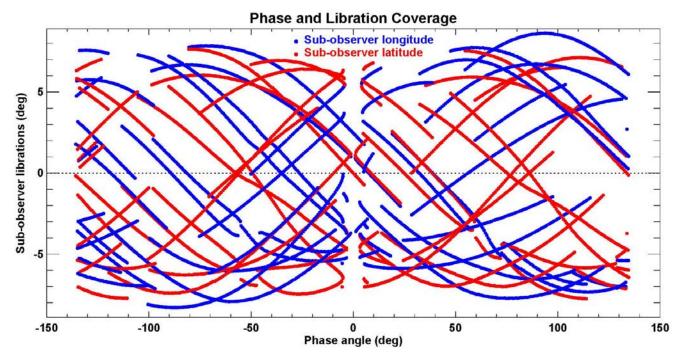
## **Moon Viewing Opportunities for CPF RS**

To build a database of lunar measurements for a calibration reference, extensive coverage of the geometric parameter space (phase angles and lunar librations) is essential.

Preliminary results from the CPF Moon view simulation:

3502 points = view opportunities with > 4 minutes duration

>80% have less than 10 minutes duration



Substantial coverage in one year!

plot shows all potential opportunities; not likely actual Moon acquisitions







## **Summary**

- CLARREO Pathfinder has the potential to acquire high-accuracy lunar spectral irradiance/disk reflectance measurements
  - useful to constrain and/or refine the lunar calibration reference
  - potential for inter-calibration of other sensors that view the Moon
- Addressing the rotating spatial sampling and its impact on lunar measurements (disk oversampling) will require post-processing of the lunar spectral images; methodology is currently under development
- Observability of the Moon from CPF location on ISS has constraints
  - some view obstructions from ISS structural components, both fixed and moveable (temporary)
  - simulation shows numerous potential view opportunities, but short time windows

The CLARREO Pathfinder team is enthusiastic about the potential to collect a lunar measurement database useful for calibration applications.







# Thank You!





