

Towards High Data Rate Hybrid RF/Optical Lunar Communication Architecture

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Background

Motivation

Lunar science and exploration is set to explode in the coming decade.

- NASA's Artemis Project will send first woman and next man to the moon by 2024 [1].
- Dozens of additional Lunar missions are planned by 2028 [2].
- Lunar missions will include human crews, rovers, smallSats, and more

These missions will require a reliable and high data rate network.

Existing Technology

RF (S, X, K band): Robust, but limited capability

- Mature and well-supported
- Low alignment requirements
- Operates under most weather conditions
- Difficult to achieve high data rates

Free Space Optical (FSO): High data rate, hard to use

- Gbps rates are common
- Stringent point/tracking requirements
- Easily disrupted by weather
- Sensitive to atmospheric turbulence [3]

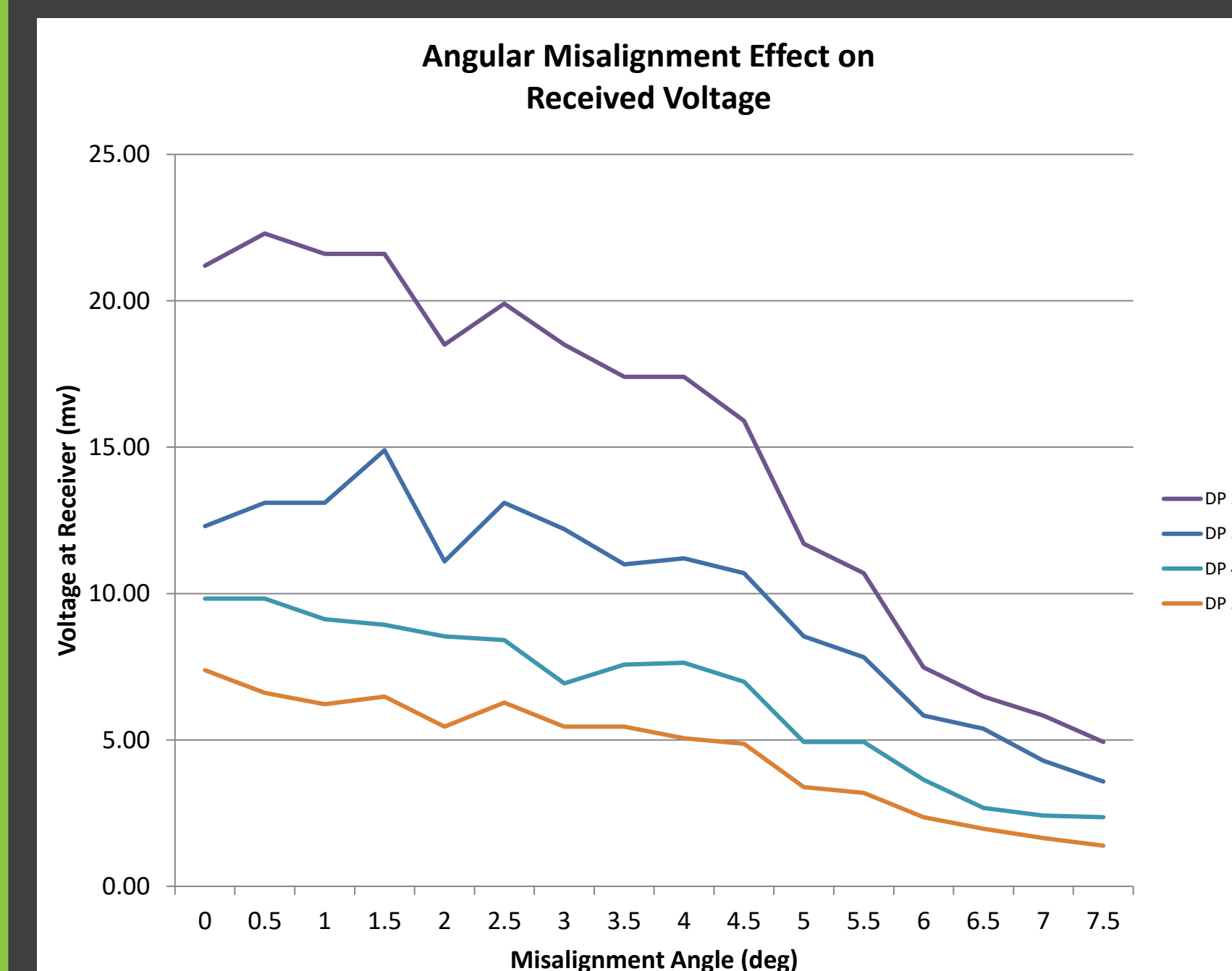
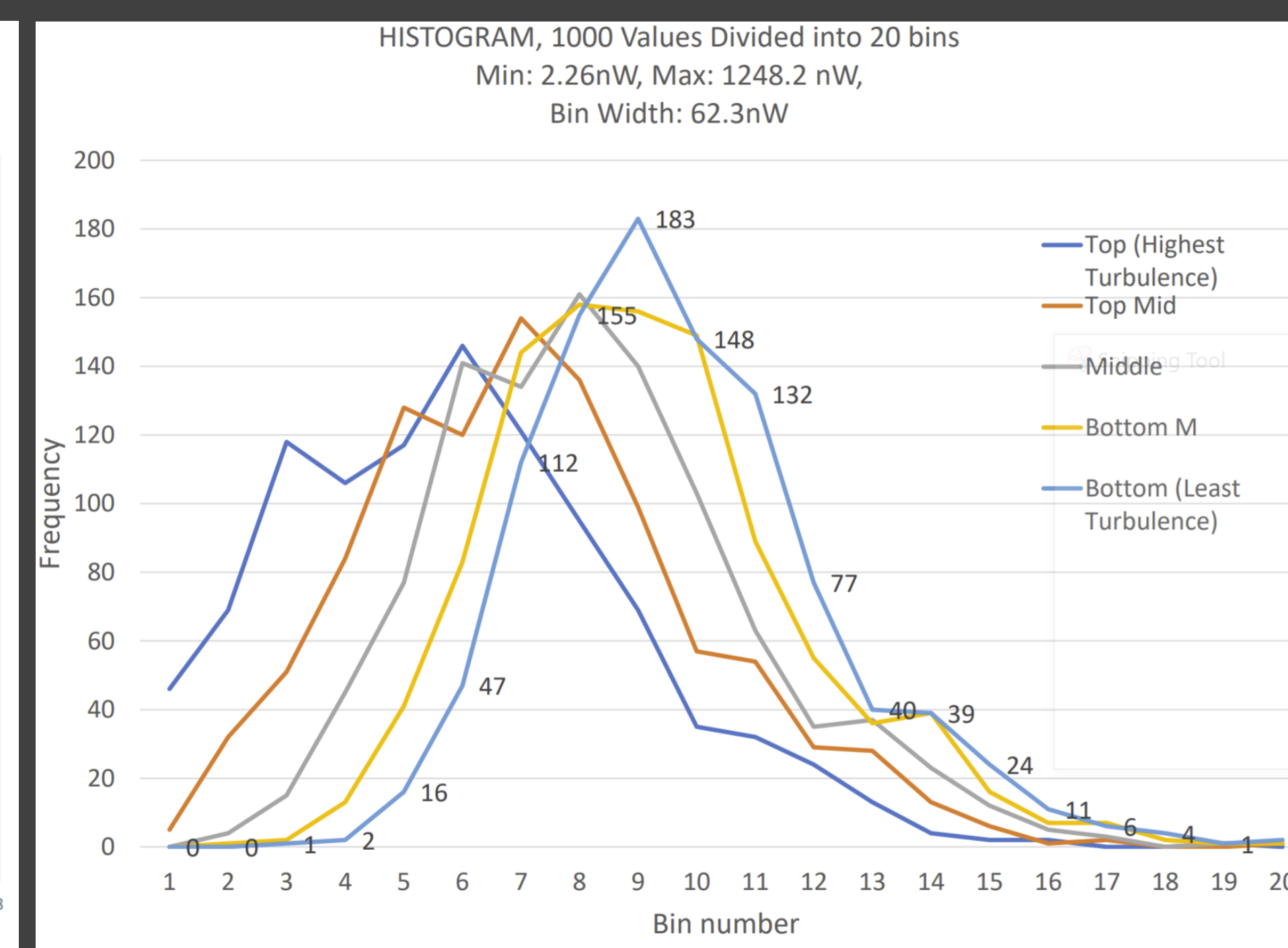
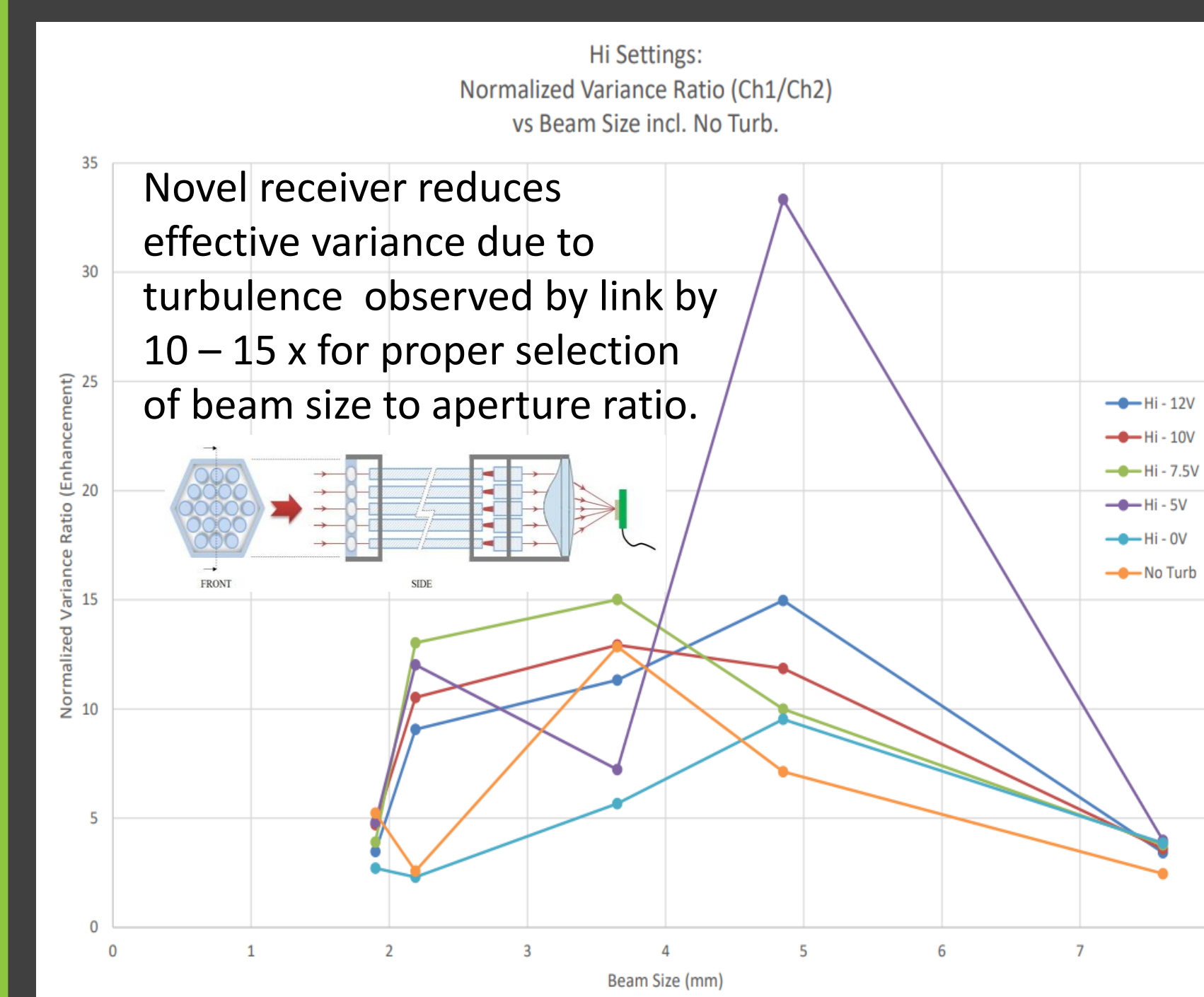
Neither can provide a network that is **both** reliable and fast.

The Solution: Hybrid Network Design

Combine the high throughput of FSO with the reliability of RF. This is a difficult engineering challenge. There are many questions:

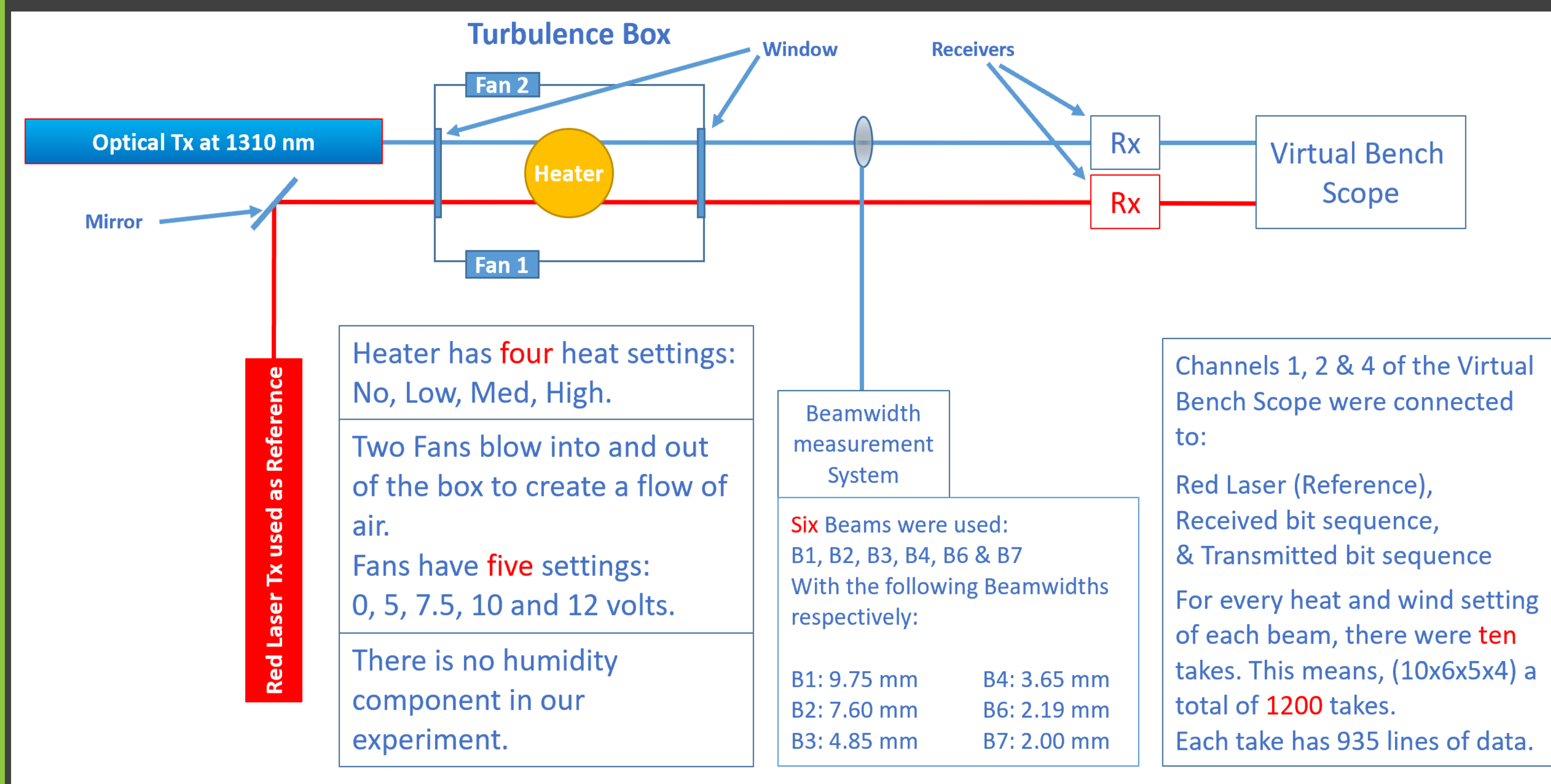
- Understanding the limits of FSO (atmosph., weather, etc.)
- Integration of RF/Optical Systems
- Network architecture and switching strategies

Active Projects: Atmospheric Effects

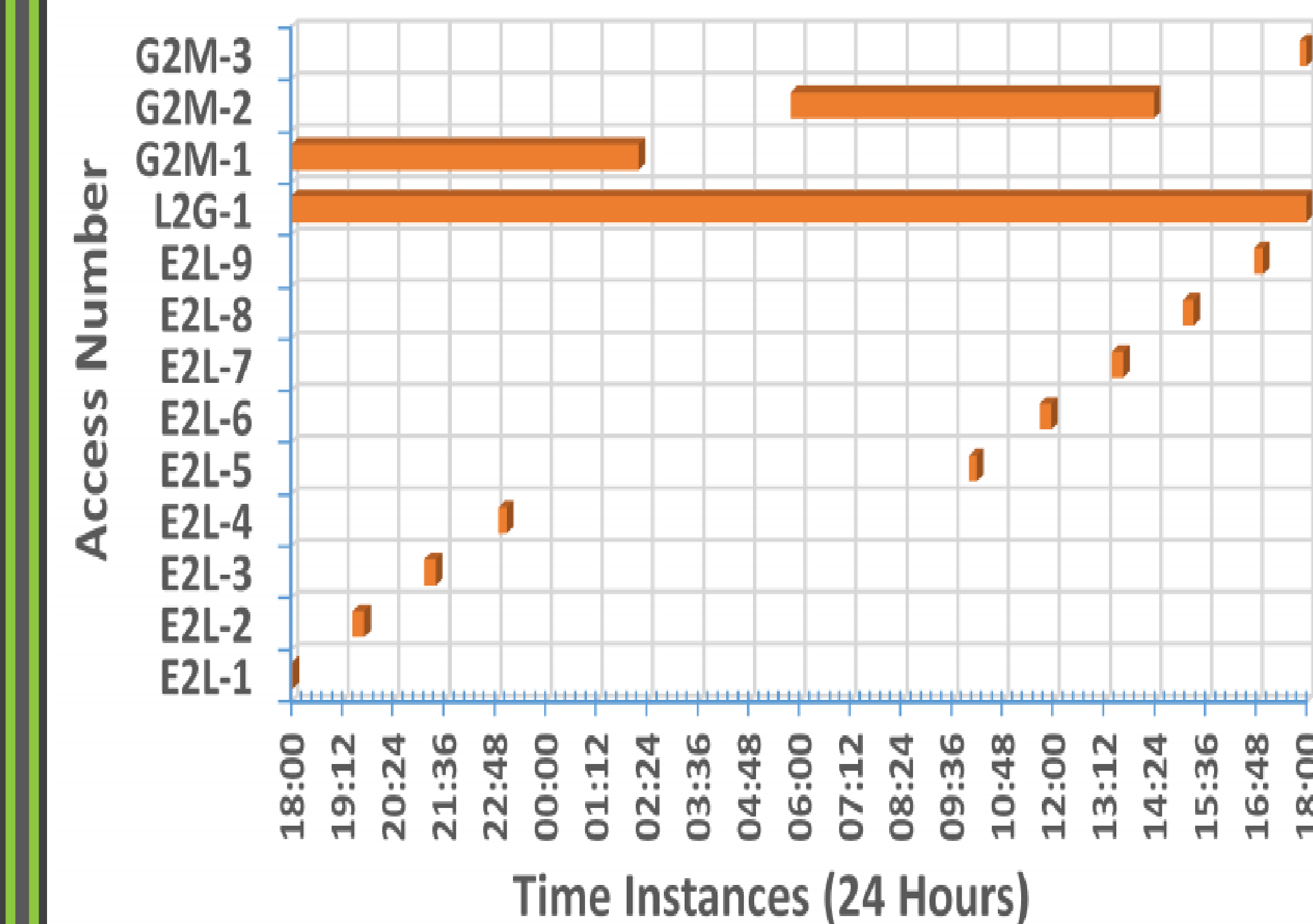
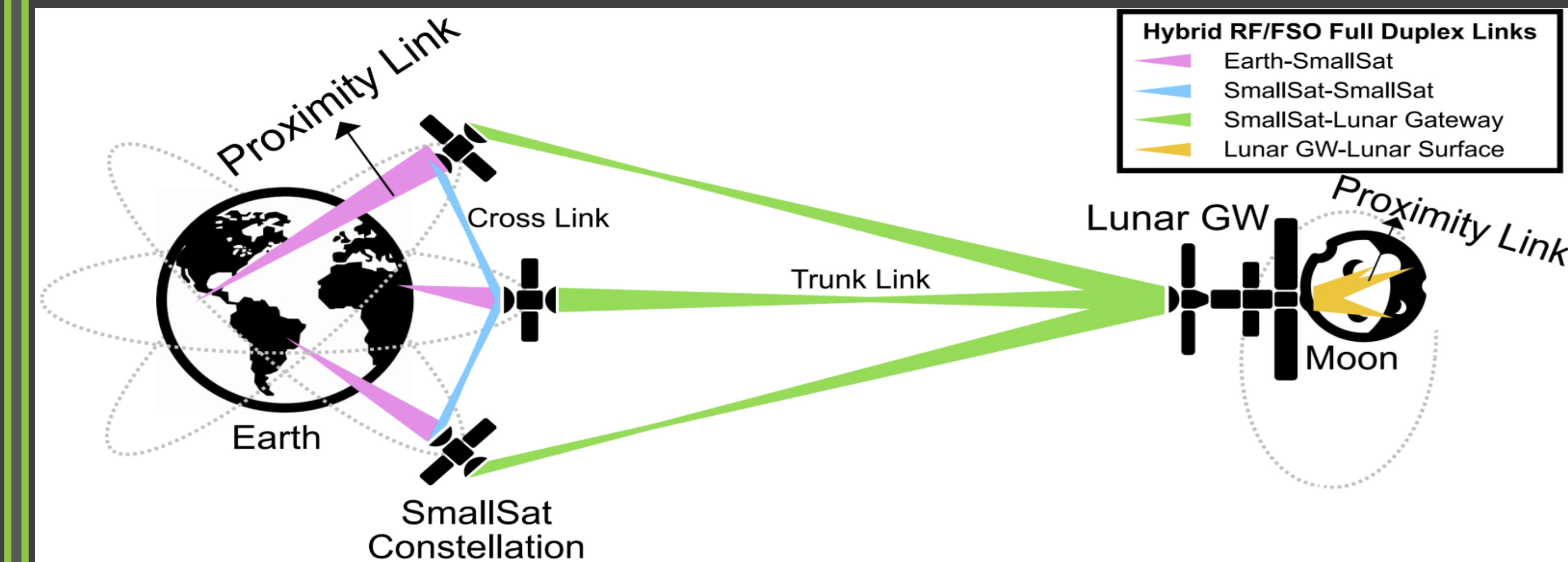


Simulated distributions of received power at satellite for 1000 km earth to LEO link, for different levels of turbulence. Six-layer atmosphere model used. Proper design of receiver – to achieve a specific minimum power and desired BER – can produce the needed link up-time to achieve a desired data throughput in a limited satellite window.

Novel Receiver has larger angular field of view – reduces aiming restrictions, could allow multiple access at same receiver.



Hybrid System Case Study



Statistical parameters of Earth to LEO access duration for different LEO constellations

LEO Const.	Access Count	Min (sec)	Mean (sec)	Max (sec)	Sum (sec)
1-by-1	09	150.628	0800.570	1026.864	07205.132
2-by-2	29	150.628	0940.415	1185.900	27272.037
3-by-3	43	499.098	1361.578	1763.173	58547.839
4-by-4	01	86400.00	86400.00	86400.00	86400.00

An experimental, indoor hybrid system is under construction (1 km path).

References

1. "Artemis overview," Available: <https://www.nasa.gov/sites/default/files/atoms/files/artemisplan-20200921.pdf>, accessed: 2021-05-04
2. "The Future Lunar Communications Architecture," Report of the Interagency Operations Advisory Group Lunar Communications Architecture Working Group, 2019. [Online]. Available: <https://www.ioag.org/Public%20Documents/Lunar%20communications%20architecture%20study%20report%20Final%2009-25-2019.docx>.
3. D. J. Israel et. al., "LunaNet: a Flexible and Extensible Lunar Exploration Communications and Navigation Infrastructure," in 2020 IEEE Aerospace Conference, 2020, pp. 1–14.
4. D. J. Israel et. al., "Laser Communications Relay Demonstration (LCRD) update and the path towards optical relay operations," 2017 IEEE Aerospace Conference, 2017, pp. 1-6, doi:
5. Integrated Radio Optical Communications (iROC), High speed beacon-less optical communications and proven radio-frequency technology.
6. T. D. Haws et. al., "SLS, the Gateway, and a Lunar Outpost in the Early 2030s," in 2019 IEEE Aerospace Conference, 2019, pp. 1–15.
7. IOAG Code and Modulation WG, "IOAG recommendations on preferred coding and modulation schemes," Interagency Operations Advisory Group, Tech. Rep. V1.1, Oct. 2019.