JAXA's Solar System Exploration Program with Small Satellites: From PROCYON and EQUULEUS to Outer Solar System Exploration

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PROCYON (2014) with U. of Tokyo

Size / Weight 55cm / 65kg

Orbit Interplanetary (Earth-resonant)

Achievements

✓ Demonstration of <u>50 kg-class deep space exploration micro-</u> <u>spacecraft bus</u> including:

• Deep space X-band communication system

- X-band GaN SSPA (>32.7%, world's highest efficiency)
- Xe-based ion / cold gas integrated propulsion system
 Science mission in deep space (Geocorona observation., etc)

EQUULEUS (2022) with U. of Tokyo

Size / Weight 6U / 10.5kg

- OrbitCislunar (Earth-Moon system)
(launched as Artemis-1 CubeSat)Achievements
- ✓ Demonstration of <u>deep space exploration CubeSat bus</u> including miniature deep space X-band transponder
- ✓ Efficient / precise trajectory control demonstration withing the Sun-Earth-Moon region
- World's first orbital maneuver beyond LEO using waterbased propulsion system

Comet Interceptor with ArkEdge Space
(planned delivery to ESA: 2026)Size / Weight24U / 35kg (one of the three S/C)OrbitSEL2 → Comet flyby trajectory

Mission

First-ever observation of a long-period comet by ESA and JAXA, which will be performed by main S/C + two small probes (one of them contributed by JAXA)
 Autonomous flyby observation by a CubeSat-class probe
 Mission assurance/reliability approach using COTS units
 Techology transfer and support for start-up company



OPENS-0: the first mission of **OPENS** (Outer Planet Exploration by Novel Small Spacecraft) program



Saturn ring fast/close flyby observation



Primary mission objective

Demonstration of key technologies to <u>enable outer planet</u> <u>exploration with low-cost small (100kg-class) spacecraft</u> <u>that can be launched by small launch vehicles</u>

LFB sequence

Key technologies to be demonstrated

Light-weight membrane-based solar array paddle
 50W@9AU, 9m², 24kg without gimbals, equivalent to >150W/kg at Earth distance

 Thermal design to minimize the required heater power in the outer planet region





(credit: NASA)

 Cost/weight effective reliability approach to survive long term deep space environments while using COTS units (e.g., hibernation operation)

Programmatic approach for higher frequency outer planet exploration in the 2030s after OPENS-0 (OPENS-1,2,...X)

Large launcher and/or orbital transfer vehicle + multiple missions (constellations of small spacecraft)



Small launcher + stand-alone independent small spacecraft



Mothership-daughtership collaboration within an international flagship mission



(image source: NASA)