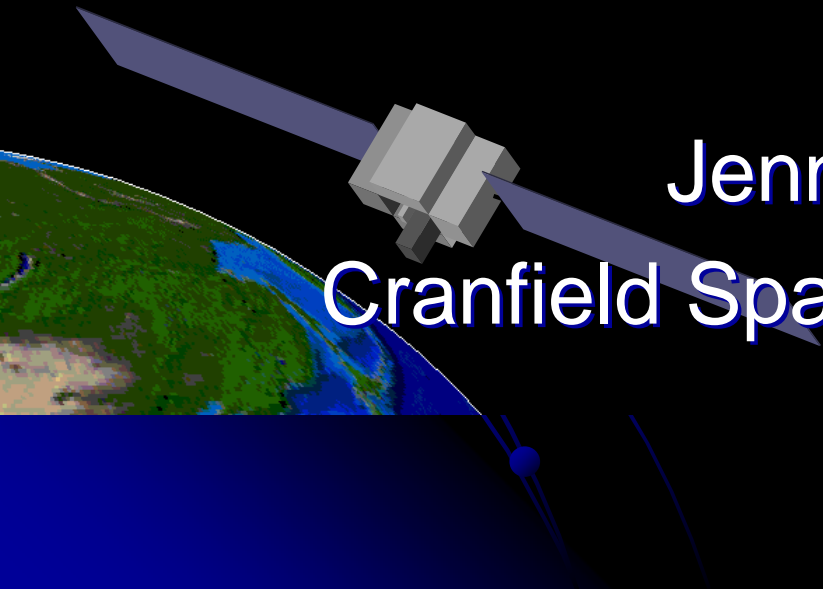


Modular Architecture & Product Platform Concepts Applied to Multipurpose Small Spacecraft

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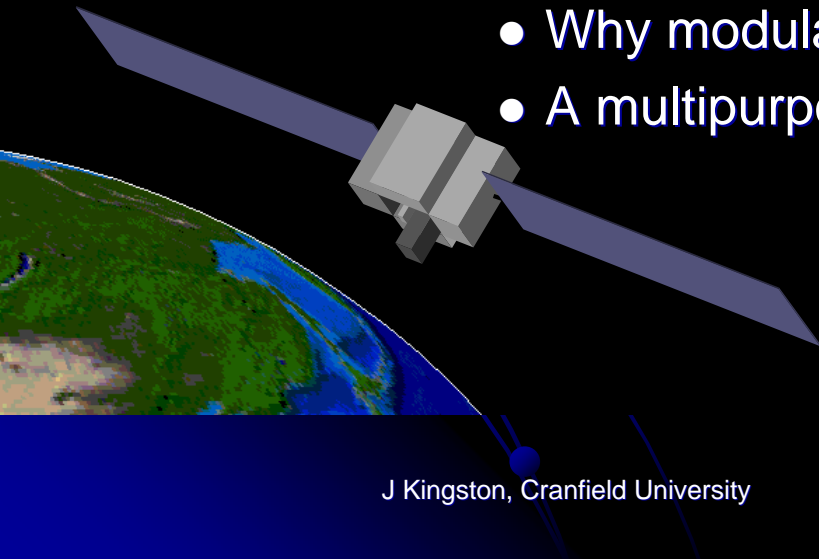
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Introduction

This presentation will introduce the concept of product platforms to enable modular architecture, and then apply it to the case of multipurpose small spacecraft

- Why multipurpose small spacecraft
- Why modularity & product platforms
- A multipurpose smallsat as a product platform

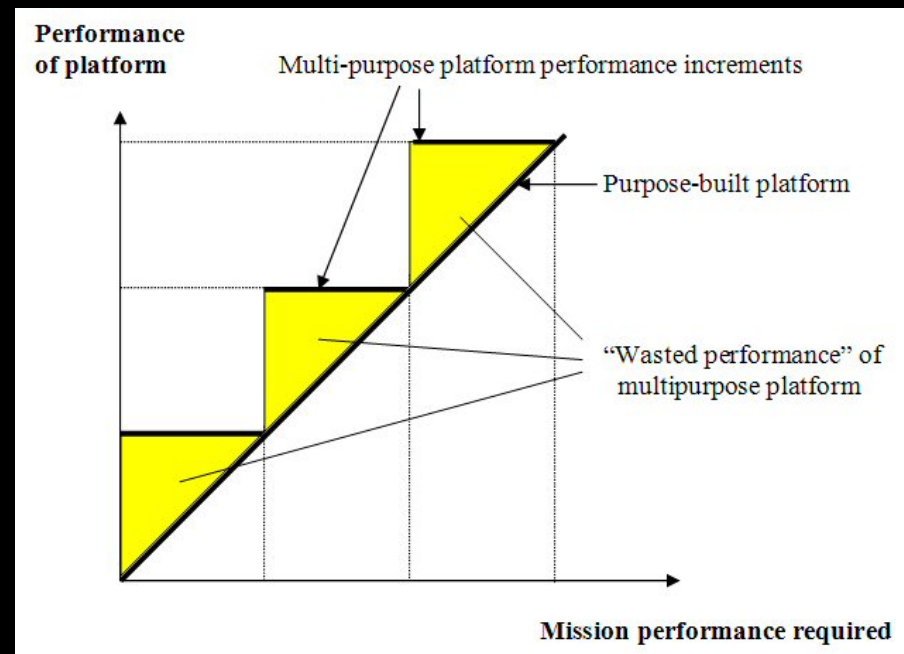


Multipurpose S/C Platforms

- Missions based on multi-purpose smallsat platforms have a number of advantages:
 - Time to flight – design around existing platform reduces Phase A/B workload, makes use of existing supply chain infrastructure, established knowledge base of team
 - Cost – likely to obtain higher performance for given cost, as R&D outlay spread over multiple missions
 - Risk – reduced risk by using proven designs and processes
- But, there are also drawbacks...

Multipurpose Platforms – The Drawbacks

- Resulting S/C unlikely to be as good a “fit” to mission specification as bespoke design
- If the multipurpose platform requires too much customisation, the benefits are lost
- Risk of “wasted performance” where standard platform models are used:

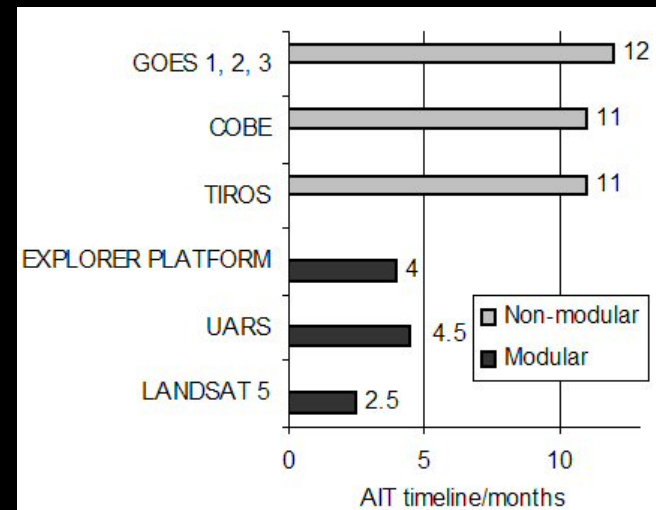


Use of Modularity

Modularity may be used to mitigate the identified drawbacks, and to enhance the benefits:

- System upgrading – modularity can ease the upgrade process, and reduce the performance increments in the products offered (reducing wasted performance)
- Integration & testing – increase concurrency in the AIT process, & use standard, simple interfaces between modules to further reduce time-to-flight

e.g. see comparison of AIT timelines for modular vs non-modular GSFC spacecraft...



Product Platforms – A Framework for Modularity

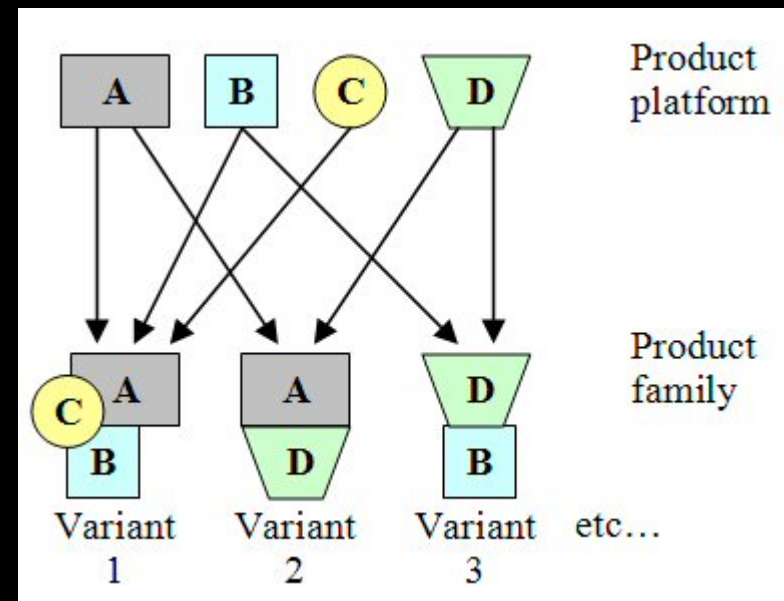
“A set of subsystems & interfaces developed to form a common structure from which a stream of derivative products is derived”*

- Strategy now in common use in electronics, automotive and consumer electrical industries
- Useful in environments where multiple, related products are offered

*Meyer, 1997

Product Platform & Product Family

- The platform is the set from which a family of different, but related, products can be produced:



Benefits of Product Platform Strategy

- In the automotive industry, Volkswagen-Audi Group produces a family of 19 vehicles based on its VW A-platform:



Audi TT



Skoda Octavia



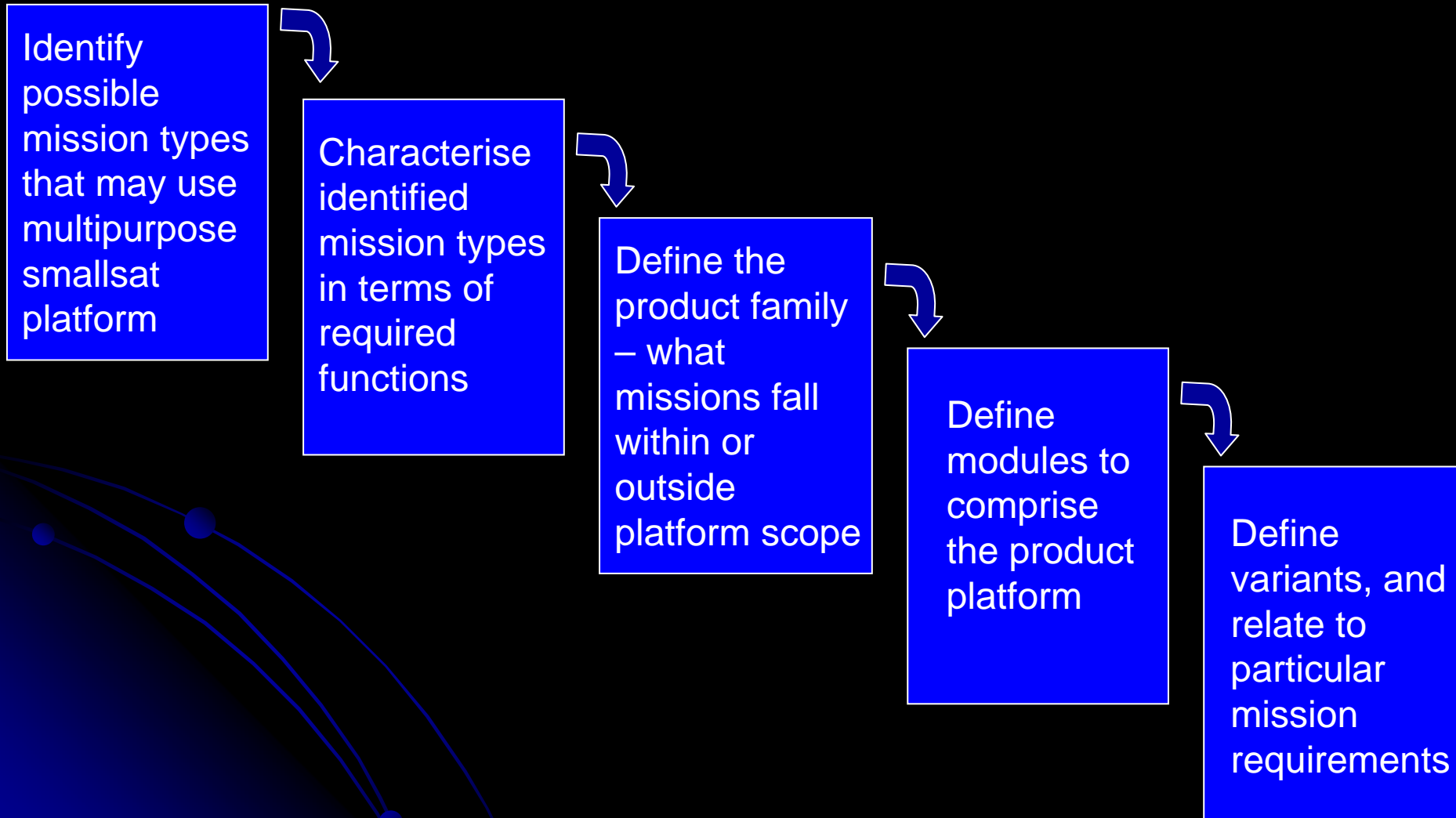
VW Golf

Savings made through use of product platforms are estimated by VW at \$1.5bn per year

Application of Product Platforms in the Space Sector

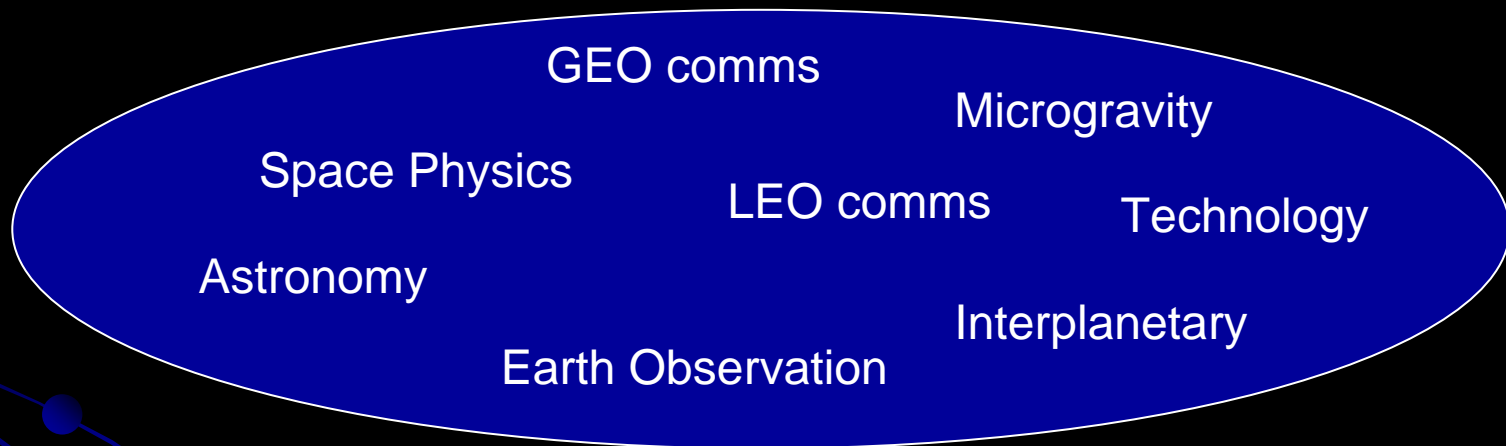
- Product platform theory evolved within the mass-customisation industries
 - Characterised by high volumes and (relatively) low unit costs
- Spacecraft manufacture is characterised by very low volumes, and very high unit costs
- However, product platform concepts have been applied in the architecture of interplanetary missions (JPL) and spacecraft communications subsystems (AeroAstro)
- The key to space application is the modular architecture at the heart of product platforms

Application to Multipurpose Smallsats



Identifying Mission Types

- Identify smallsat missions, & group into types by characterisation of typical functional requirements:



- Each of these mission types can be thought of as a potential variant in the spacecraft product family
- Actual product family is likely to be a subset of these

Characterising Mission Types

Mission types were characterised by reference to the following parameters:

- Physical accommodation of payload(s) – no., size, geometry, mass, field of view
- Communications & data handling – data rates, storage, operations, processing
- Attitude – pointing & stabilisation type, accuracy, stability, manoeuvrability
- Orbit – type, maintenance requirement, position knowledge
- Power – mean & peak levels, duty cycles
- Environmental – thermal, radiation, cleanliness etc

Variant-Function Matrix

- This identifies the required functions for each variant:

| Variants | Functions | | | | | | | | | | | | | | | | | | | | | | | | | | | | |
|-------------------|------------------------|-------------|----------------------|--------------|----------------------------|------------------|---------------|--------------------|-----------------|---------------|---------------|-----------|----------|---------|--------------------------|-------------------------|---------------|-------------------------------|---------------------|---------------------|--------------|--------|-------------------------|-------|----------------------------|-------------------|-----------------------|------------------------|---------------------|
| | Attitude Determination | | | | Attitude Control | | | | Propulsion | | | | Power | | | | Data Handling | | | Comms | | | Structures & Mechanisms | | | | | | |
| | Earth sensing | Sun sensing | Inertial referencing | Star sensing | Magnetic field measurement | Reaction control | Momentum bias | Spin stabilisation | Magnetic torque | Gas thrusters | Kick motor(s) | Hydrazine | Electric | Other † | Electrical power storage | Solar energy conversion | Power control | Radioisotope power generation | S/C data processing | P/L data processing | Data storage | S-Band | X-Band | TDRSS | Primary structural support | Solar array drive | Mechanical deployment | Extend boom structures | Re-entry/ retrieval |
| Astronomy | | X | X | X | | X | X | | X | X | X | | | | X | X | X | | X | X | X | | X | X | X | | X | | |
| Space physics | X | X | | | X | | | X | | X | | | | | X | X | X | | X | | X | X | | | X | | X | X | |
| Interplanetary | | X | X | X | X | X | | X | | X | X | X | X | X | X | X | X | X | X | X | X | X | X | | X | X | X | X | X |
| Microgravity | X | X | X | | | X | | | X | | | | | | X | X | X | | X | | X | X | X | X | X | | X | | X |
| Earth Observation | X | X | | X | X | X | X | | X | X | | X | | | X | X | X | | X | X | X | X | X | X | X | X | X | X | |
| LEO Comms | X | X | | | X | X | X | | X | X | | X | | | X | X | X | | X | | X | X | | | X | X | | | |
| GEO Comms | X | X | | | | X | X | | | X | X | X | | | X | X | X | | X | | X | X | | | X | X | X | | |

† e.g. solar sailing

Defining the Product Family

- In the ideal case, a multipurpose S/C may be designed around a product platform that gives a variant for any mission type
- In reality, we see “sparse functions” in the variant-function matrix e.g. re-entry provision, RTG power
- These functions would dilute the product platform with seldom-used modules
- Modules that are unique to a product, and have a key impact on the final product, have been described as *form defining modules* *
- For spacecraft, a major form-defining module is the payload...

*Chandrasekaran, 2004

Defining the Product Family

- The XMM payload: a form defining module...



Defining the Product Family

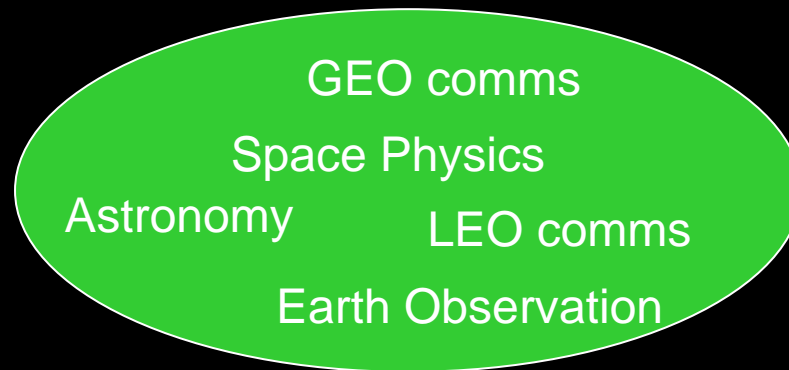
- It is clear that products as complex as spacecraft cannot be entirely formed from a standardised set of modules
- To define a suitable scope for the product platform, we must strike a balance of:

Maximising range of applicable missions

Minimising occurrence of form-defining modules

Defining the Product Family & The Product Platform Modules

- Examination of clustering of functional requirement types suggests that the product platform should provide variants supporting this mission set:



Microgravity
Technology
Interplanetary

- Modules showing highest frequency of occurrence in the product-variant matrix are chosen to form the product platform
- Specific requirements for the modules are quantified by allocating “capability increments”, based on the parameters of past & current missions

Modules & Variants

- A variant-module matrix can be defined - indicating variants suitable for different mission types, and their constituent modules:

| Product Platform Modules ↓ | Product Family Variants → | | | | |
|---|---------------------------|---------------|-------------------|-----------|-----------|
| | Astronomy | Space physics | Earth Observation | LEO Comms | GEO Comms |
| Structural Modules (number of modules used indicated) | 4 | 3 | 4 | 4 | 4 |
| Basic Attitude Control Module (small wheels, magnetorquers) | | | | X | X |
| Enhanced Attitude Control Module (large wheels, magnetorquers) | X | | X | | |
| Basic Attitude Determination Module (Earth sensor, magnetometer, sun sensors) | | X | X | X | X |
| High accuracy Attitude Determination Module (Star tracker, IRU, sun sensors) | X | | | | |
| Cold gas Propulsion Module | X | X | | | |
| Hydrazine Propulsion Module | | | X | | X |
| Kick motor | | X | | | X |
| Data Handling Module | X | X | X | X | X |
| Payload Data Handling Module | | | X | | |
| Mass Memory Module | X | X | | | |
| S-Band Comms Module | X | X | X | X | X |
| X-Band Comms Module | X | | | | |
| TDRSS Comms Module | | | X | | |
| Battery Module (number of modules used indicated) | 2 | 1 | 4 | 4 | 5 |
| Power Control & Distribution Module | X | X | X | X | X |
| Body-mounted Solar Array Module | | X | | | |
| Deployed-fixed Solar Array Module | X | | | | |
| Deployed-articulated Solar Array Module | | | X | X | X |

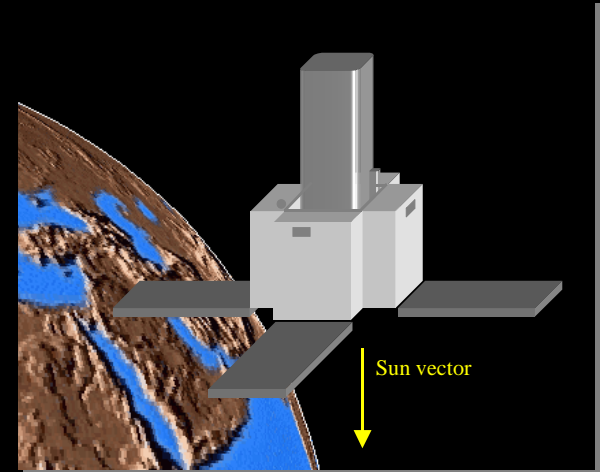
Example – Selecting a Variant

- X-ray astronomy mission used for selection of a suitable variant
- Mission parameters derived from HESSI & XTE missions
 - Payload mass 120kg, power 110W
 - Payload size 1.4m x 0.45m diameter
 - 25 arcsec pointing accuracy
 - 0.1° per sec manoeuvres
 - 10Gbits data rate over 10 min observation period
 - Inertial attitude
- Platform modules are selected to satisfy mission requirements...

Variant Selection

Key choices in selecting a suitable variant:

- Structural module configuration of 4 modules accommodates large payload instrument
- X-band comms module supports high downlink rate, S-band module for backup & TT&C
- Additional mass-memory modules provide necessary data storage
- High-accuracy attitude determination module, with enhanced attitude control module to provide necessary pointing accuracy and stability.
- High manoeuvrability provided by cold-gas propulsion module



Summary

- Product platform approach allows reduced programmatic costs and schedules
- It requires greater investment and manpower to implement, but this can be recouped in subsequent projects
- This high initial investment means it is not suitable for all suppliers
- However, the product platform approach is applicable at many levels of the spacecraft (component, subsystem, system), & may be implemented by adaptation of existing products
- It may also provide a framework for cooperation within the supply-chain